

# **ACADEMIC CURRICULA**

## **Professional Core Courses**

**AEROSPACE ENGINEERING**

**Regulations - 2018**

**SRM INSTITUTE OF SCIENCE AND TECHNOLOGY**

**(Deemed to be University u/s 3 of UGC Act, 1956)**

**Kattankulathur, Kancheepuram, Tamil Nadu, India**

Course Code	18ASC101T	Course Name	APPLIED ENGINEERING MECHANICS	Course Category	C	Professional Core	L	T	P	C
							3	1	0	4

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards	Nil		

Course Learning Rationale (CLR):	The purpose of learning this course is to:	Learning	Program Learning Outcomes (PLO)
CLR-1 :	Utilize the concept of equilibrium of particles and rigid bodies	1	1
CLR-2 :	Utilize the concept of finding centroid of planar figures and moment of inertia about different axes	2	2
CLR-3 :	Utilize with the dynamics of particles	3	3
CLR-4 :	Utilize with the dynamics of rigid bodies	4	4
CLR-5 :	Apply the concepts of mechanics to solve problems related to space mechanics	5	5
CLR-6 :	Utilize the concepts in better understanding of systems dealing with forces	6	6
Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Engineering Knowledge
CLO-1 :	Determine the forces under equilibrium	2	H
CLO-2 :	Identify the centroids and determine moment of inertia	2	H
CLO-3 :	Determine the forces acting on particle both kinetics and kinematics	2	H
CLO-4 :	Determine the forces acting on rigid body both kinetics and kinematics	2	H
CLO-5 :	Application of determining space orbit	2	H
CLO-6 :	Apply the concepts of fundamental mechanics and space mechanics in real time applications	2	H
		Expected Proficiency (%)	Problem Analysis
		Expected Attainment (%)	Design & Development
			Analysis, Design, Research
			Modern Tool Usage
			Society & Culture
			Environment & Sustainability
			Ethics
			Individual & Team Work
			Communication
			Project Mgt. & Finance
			Life Long Learning
			PSO - 1
			PSO - 2
			PSO - 3

Duration (hour)	12	12	12	12	12
S-1	SLO-1	Fundamentals of mechanics- Classification of forces, Laws of mechanics.	Determination of centroids by integration, centroids of lines, areas and volumes.	Rectilinear motion-Uniform motion and Rectangular components of velocity	Kinematics of rigid bodies
	SLO-2	Vector and vector operations problems	Determination of centroids by integration, centroids of areas	Rectilinear motion - Uniformly accelerated motion	Kinematics of rigid bodies: Linear translational motion
S-2	SLO-1	Forces on particles in 2 D- Triangular law of forces & parallelogram law of forces	Determination of centroids in composite areas	Curvilinear motion-Normal and tangential components	Kinematics of rigid bodies: Fixed axis rotation
	SLO-2	Equilibrium on particles in 2D, Lami's Theorem, Free body diagram	Centroids of volumes, Centre of gravity	Curvilinear motion- Normal and tangential components	Kinematics of rigid bodies: Fixed axis rotation
S-3	SLO-1	Action & Reaction, Equilibrium on particles in 2 D – Equations of Equilibrium	Pappus guildinus Theorem I	Curvilinear motion- Radial and transverse components	Kinematics of rigid bodies: relation between linear and rotation
	SLO-2	Forces in space	Pappus guildinus Theorem II	Curvilinear motion- Radial and transverse components	Kinematics of rigid bodies: relation between linear and rotation
S-4	SLO-1	Solving Problems	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Solving Problems	Solving Problems	Solving Problems
S-5	SLO-1	Statics of rigid body in 2D – Moment & Varignon's Theorem	Moment of inertia	Cylindrical coordinates- Newtons second law, D'Alembert's principle.	General plane motion-Absolute and relative velocity in plane motion
	SLO-2	Statics of rigid body in 2D – Force Couple System	Determination of moment of inertia by Integration	Cylindrical coordinates- Newtons second law, D'Alembert's principle.	General plane motion-Absolute and relative velocity in plane motion

S-6	SLO-1	Equilibrium of Rigid bodies in 2D	Parallel axis theorem	Principle of work and energy	General plane motion: Crank- Rod Mechanism	Newton's Law of Gravitation
	SLO-2	Equilibrium of Rigid bodies in 2D : Support Reactions, Types of Support	Parallel axis theorem	Principle of work and energy	General plane motion: Crank- Rod Mechanism	Sample problems: Conservation of angular momentum and newton's law of gravitation
S-7	SLO-1	Analytical method to determine the support reactions of beam	Perpendicular axis theorem	Principle of impulse and Momentum.	Instantaneous centre of rotation in plane motion	Sample problems: Conservation of angular momentum and newton's law of gravitation
	SLO-2	Moment of Uniformly varying loads	Perpendicular axis theorem	Principle of impulse and Momentum.	Instantaneous centre of rotation in plane motion	Sample problems: Conservation of angular momentum and newton's law of gravitation
S-8	SLO-1	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
S-9	SLO-1	Truss: Classification, perfect/Imperfect frame, Analysis of perfect frame	Polar moment of inertia,	Impact of Elastic bodies	D'Alembert's principle : Linear motion	Trajectory of a particle under a central force
	SLO-2	Determine the support Reaction in truss	Radius of gyration	Derivation of Elastic coefficient	D'Alembert's principle : Rotation motion	Trajectory of a particle under a central force: Application to space mechanics
S-10	SLO-1	Analysis of perfect Frame by method of joints: Simply supported	Mass moment of inertia of solid objects	Impact of Elastic bodies-Direct central	Principle of work and energy for a rigid body : Linear motion	Kepler Law of planetary motion
	SLO-2	Analysis of perfect Frame by method of joints: Cantilever	Mass moment of inertia of solid objects	Impact of Elastic bodies-Direct central	Principle of work and energy for a rigid body: Rotation motion	Sample problems: Central force
S-11	SLO-1	Analysis of perfect Frame by method of sections: Simply supported	Mass Moment of inertia of thin plates	Impact of Elastic bodies- Oblique central impact.	Principle of impulse, momentum for plane motion of a rigid body: Linear motion	Sample problems: Central force
	SLO-2	Analysis of perfect Frame by method of sections: Cantilever	Mass Moment of inertia of thin plates	Impact of Elastic bodies- Oblique central impact.	Principle of impulse, momentum for plane motion of a rigid body: Rotation motion	Sample problems: Periodic time
S-12	SLO-1	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems

Learning Resources	1. Ferdinand P. Beer, E. Russell Johnston Jr., David Mazurek, Philip J Cornwell, Vector Mechanics for Engineers: Statics and Dynamics, 10 <sup>th</sup> ed., McGraw Hill, 2013	3. NPTEL Engineering Mechanics Lectures by IIT Guwahati 'https://nptel.ac.in/courses/112103109/'
	2. Shames, I.H., Krishna Mohana Rao, G., Engineering Mechanics (Statics and Dynamics), Dorling Kindersley (India) Pvt. Ltd. (Pearson Education), 2006	

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry		Experts from Higher Technical Institutions
1. Dr. S. Raja, National Aerospace Laboratories, Bangalore, raja@nal.res.in		1. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu
2. Dr. Vinay Kumar Gupta, National Physical Laboratory, guptavinay@nplindia.org		2. Dr. S. Elangovan, BIHER, Chennai, email : subelango@yahoo.co.in
		Internal Experts
		1. Mr. K. B. Ravichandrakumar, SRMIST
		2. Mr. lynthezhuthon, SRMIST

Course Code	18ASC102J	Course Name	APPLIED FLUID MECHANICS	Course Category	C	Professional Core	L	T	P	C
							3	0	2	4

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):	The purpose of learning this course is to:
CLR-1 :	Identify the characteristics of fluids and utilize the pressure measuring devices
CLR-2 :	Solve the basic fluid flow problems and apply the system and control volume concept in various fluid flow problems
CLR-3 :	Identify the mathematical techniques of potential flow problems
CLR-4 :	Solve the basic dimensional analysis and fluid flow through pipes
CLR-5 :	Analyze the basic concepts of boundary layer in fluid flow
CLR-6 :	Explore advanced level of fluid mechanics applications

Learning		
1	2	3
Thinking (Bloom)	Efficiency (%)	Attainment (%)

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Knowledge	Analysis	Development	Design, Research	Usage	Culture	& Sustainability		Team Work	Communication	Finance & Economics	Learning			

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLO-1 :	Accrue the knowledge of fluid properties and principle and function of pressure measuring instruments	2	85	75	H	M	M	L	-	-	-	-	L	-	-	M	L	-	-
CLO-2 :	Analyze the fluid flow problems and system and control volume concept	2	85	75	H	H	M	M	-	-	-	-	L	-	-	M	L	-	-
CLO-3 :	Apply the mathematical techniques of potential flow problems	2	85	75	H	H	M	M	-	-	-	-	-	-	-	M	-	-	-
CLO-4 :	Apply the dimensional analysis and fluid flow through pipes	2	85	75	H	H	M	M	-	-	-	-	L	-	-	M	L	-	-
CLO-5 :	Accrue the knowledge about boundary layer concept	2	85	75	H	M	M	L	-	-	-	-	-	-	-	M	-	-	-
CLO-6 :	Accrue comprehensive knowledge in fluid mechanics applications	2	85	75	H	H	M	M	-	-	-	-	L	-	-	M	L	-	-

Duration (hour)	15	15	15	15	15
S-1	SLO-1 Introduction to fluid mechanics	Lagrangian and Eulerian description of fluid flow	Pitot – tube	Dimensional Analysis	Pipe friction major and Minor losses
	SLO-2 Brief history of fluid mechanics	Types of fluid flow, streamlines, path lines, and streak lines.	Numerical problems	Rayleigh's method, numerical problems	Numerical problems
S-2	SLO-1 Fluids and their properties	System and Control volume concept	Introduction to potential flow	Buckingham's Pi – theorem	Numerical Problems in parallel,
	SLO-2 Density, viscosity, surface tension	Introduction to Reynolds transport theorem	Equation of streamline	Buckingham's Pi – theorem procedure	Series and branched pipes.
S-3	SLO-1 Properties of fluids numerical problems	Reynolds transport theorem	Stream function, Velocity potential function	Numerical problems on Buckingham's Pi – theorem	Boundary layer theory introduction
	SLO-2 Compressibility and bulk modulus	Reynolds transport theorem and its applications	Basic elementary flows	Numerical problems	Fluid flow over bodies
S 4-5	SLO-1 Lab 1: Determine coefficient of discharge of orifice meter	Lab 4: Repeat class	Lab 7: Performance test on radial centrifugal air blower	Lab 10: Performance test on forward centrifugal air blower	Lab 13: Major loss due to friction in pipe flow
S-6	SLO-1 Fluid statics-Pascal's law	Reynolds transport theorem, applications in finite control volume analysis	Uniform parallel flow stream function and velocity potential function	Applications of important dimensionless numbers	Flow over a flat plate
	SLO-2 Numerical problems on Pascal's law	control volume analysis Numerical problems	Source flow and sink flow stream function and velocity potential function	Numerical problems	Boundary layer development on a flat plate
S-7	SLO-1 Hydrostatic law	Euler's equation of motion along a streamline	Free vortex	Flow through pipes	Displacement thickness, momentum thickness



	SLO-2	Piezometric head, and Numerical problems	Bernoulli's equation	Free vortex stream function and velocity potential function	Laminar and turbulent flow	Energy thickness
S-8	SLO-1	Manometry- simple manometer	Bernoulli's equation - Numerical problems	Forced vortex	Hagen - Poiseuille flow in circular pipes	Numerical problems on
	SLO-2	Numerical problems on simple manometers	Numerical problems on Bernoulli's equation	Combination of elementary flows	Hagen – Poiseuille equation	displacement thickness, momentum thickness, Energy thickness
S 9-10	SLO-1	Lab 2: Determine coefficient of discharge of venturimeter	Lab 5: Determine Impact force of water jet on vane	Lab 8: Repeat class	Lab 11: Determine type of flow by Reynolds apparatus	Lab 14: Performance test on reciprocating air compressor
	SLO-2	Differential manometer- U-tube differential manometer	Bernoulli's equation – Application venturimeter, orifice meter, pitot tube	Doublet flow	Hagen – Poiseuille equation applications	Drag on a flat plate
S-11	SLO-1	Differential manometer- Inverted U-tube differential manometer	Numerical problems on Venturimeter	Non-lifting flow over a cylinder	Development of flow in pipes Darcy-Weisbach equation	Separation of flow over bodies, streamlined and bluff bodies
	SLO-2	Numerical problems on Inverted U-tube differential manometer	Numerical problems on Venturimeter	Pressure and velocity distributions	Pipe friction	Lift and Drag on cylinder
S-13	SLO-1	Inclined manometer	Orifice meter	Lifting flow over a cylinder	Numerical problems on Darcy-Weisbach equation	Lift and Drag on Aerofoil
	SLO-2	Numerical problems on Inclined manometer	Orifice meter discharge equation, and numerical problems	pressure and velocity distributions	Numerical problems on Pipe friction	Lift and Drag on cylinder and Aerofoil Numerical problems
S 14-15	SLO-1	Lab 3: Verify Bernoulli's theorem	Lab 6: Minor losses due to pipe fittings in pipes	Lab 9: Performance test on backward centrifugal blower	Lab 12: Repeat class	Lab 15: Repeat class
	SLO-2					

Learning Resources	1. Kumar, K.L., Engineering Fluid Mechanics, 8 <sup>th</sup> ed., S. Chand, New Delhi, 2016	3. Irving H. Shames, Mechanics of Fluids, 4 <sup>th</sup> ed., McGraw-Hill, 2003
	2. Munson, Bruce R., Young, Donald F., Okiishi, Theodore H., Huebsch, Wade W. Fundamentals of Fluid Mechanics, 7 <sup>th</sup> ed., John Wiley & Sons, Inc. 2016	4. Streeter, Victor, Bedford, K.W., Wylie, E. Benjamin, Fluid Mechanics, 2 <sup>nd</sup> ed., Tata McGraw Hill, New Delhi, 1997

#### Learning Assessment

	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember Understand	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
Level 2	Apply Analyze	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
Level 3	Evaluate Create	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Total	100 %		100 %		100 %		100 %		100%	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. S. Raja, National Aerospace Laboratories, Bangalore, raja@nal.res.in	1. Dr. S. Elangovan, BIHER, Chennai, email : subelango@yahoo.co.in	1. Mr. S. Rajkumar, SRMIST
2. Dr. Vinay Kumar Gupta, National Physical Laboratory, guptavinay@nplindia.org	2. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	2. Mr. M. Abdur Rasheed, SRMIST

Course Code	18ASC103T	Course Name	AERO ENGINEERING THERMODYNAMICS	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1 :	Identify the engineering and practical applications of Heat, Energy and Work			
CLR-2 :	Identify the applications of Thermodynamics on Engineering systems			
CLR-3 :	Identify the significance of Thermodynamic Laws			
CLR-4 :	Create insights to the concepts of Entropy and Exergy			
CLR-5 :	Analyze the working principle of Heat Energy driven systems			
CLR-6 :	Utilize the Thermodynamic concepts in physics for the broad understanding of engineering and technology			

Learning		
1	2	3
Learning (Bloom)	Proficiency (%)	Attainment (%)

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLO-1 :	Identify the laws of Thermodynamics and its applications to Aerospace Engineering	2	80	70	H	M	L	L	-	-	-	-	-	-	-	-	H	M	M
CLO-2 :	Comprehend the concept and applications of energy, entropy and exergy	2	80	70	H	M	M	M	-	-	L	-	-	L	-	-	H	M	M
CLO-3 :	Understand various gas and vapor power cycles with applications	2	80	70	H	M	L	M	-	-	-	-	L	-	-	-	H	-	-
CLO-4 :	Understand the gas mixture behavior and chemical reactions	2	80	70	H	M	M	M	-	-	M	M	M	M	L	H	M	M	M
CLO-5 :	Utilize the fundamental concepts for the physical understanding of engineering and technology	2	80	70	H	M	M	M	M	L	L	-	L	M	L	H	M	M	M
CLO-6 :	Apply the Thermodynamic Principles to Aerospace Engineering Applications	2	80	70	H	M	M	M	M	L	L	M	L	M	L	H	M	M	M

Duration (hour)		9	9	9	9	9
S-1	SLO-1	Basic Concepts: Microscopic,macroscopic point of view, Path and point functions.	Limitations of first law of Thermodynamics. Introduction to Heat Reservoirs, Sources and Sinks	Limitations of Second Law of Thermodynamics	Role of Carnot cycle in Aerospace engineering	Mass fraction and mole fractions
	SLO-2	Intensive and extensive, total and specific quantities.	Heat Engine, Refrigerator, and Heat pump. Thermal efficiency of heat engines.	Explanation of the Concept of Entropy	Introduction to Otto cycle, Diesel cycle, Dual cycle	p-v-t behavior and properties of ideal gas mixtures
S-2	SLO-1	System and types. Zeroth law of thermodynamics, Thermodynamic equilibrium	Second law of Thermodynamics: C.O.P, Kelvin-Planck statement	Clausius inequality, T-s diagram	Indicator diagram Mean effective pressure	Dalton's law of partial pressures, Avogadro's law
	SLO-2	Quasi-static, reversible and irreversible processes. Heat and work transfer, sign convention	Clausius statement of second law and equivalence of statements.	Entropy change for different processes.	Comparison of Otto, Diesel and Dual cycles, Air standard efficiency	Gibbs-Dalton law, enthalpy and specific heat of a gas mixtures
S-3	SLO-1	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
S-4	SLO-1	First law of Thermodynamics: First law for a closed system undergoing a cycle, concept of Internal energy, change of state	Reversible and irreversible processes-causes of irreversibility	Principle of increase of Entropy, Maxwell relations, T-ds Equations, Difference and ratio of heat capacities	Introduction to Aerospace Propulsion - Brayton cycle	Chemical reactions, Combustion, Stoichiometric coefficients, Air-Fuel ratio, Equivalence ratio
	SLO-2	Energy and Work Transfer in closed	Carnot Theorem and corollary	Energy equation, Joule Thomson	Effect of Reheat, Regeneration and	Combustion and Dissociation

		systems, P-V diagram, PMM1		Coefficient, Clausius-Clapeyron equation	Intercooling	
S-5	SLO-1	Solving Problems	Absolute Thermodynamic Temperature scale	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Carnot cycle and Performance	Solving Problems	Solving Problems	Solving Problems
S-6	SLO-1	First law for an Open system: Conservation of mass, energy, steady flow energy equation	Solving Problems	Entropy change of Ideal and Real gases	Turbine and Compressor efficiency	Aerospace Chemical Propulsion: Fuels in combustion
	SLO-2	Aerospace applications of SFEE to Nozzles, Diffusers	Solving Problems	Isentropic efficiencies of Aerospace steady flow devices	Factors affecting efficiency	Enthalpy of reaction, formation and combustion
S-7	SLO-1	Cases of turbine, compressor, boiler, pump	Engineering and Practical Applications of Second Law	Exergy in Aerospace Engineering: High and low-grade energy	Equivalent Carnot cycles: Stirling and Ericsson cycle, Humphrey cycle	Gravimetric and volumetric analysis
	SLO-2	Heat exchanger and Throttling process	Aerospace Engineering Applications of Second Law	Available and non-available energy of a source and finite body	Interactive session with demo on practical working of Gas Power based Engines	Introduction to adiabatic flame temperature
S-8	SLO-1	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
	SLO-2	Solving Problems	Solving Problems	Solving Problems	Solving Problems	Solving Problems
S-9	SLO-1	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.
	SLO-2	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.	Chapter Doubt clarification.

Learning Resources	1. Nag, P. K, Engineering Thermodynamics, 6 <sup>th</sup> ed., Tata McGraw Hill, 2017 2. Rathakrishnan. E, Fundamentals of Engineering Thermodynamics, Prentice-Hall, India, 2005 3. Holman, J. P., Thermodynamics, 4 <sup>th</sup> ed., Tata McGraw Hill, 2015 4. Rayner Joel, Basic Engineering Thermodynamics, 5 <sup>th</sup> ed., Addison Wesley, 2016	5. Michael Moran, J., Howard Shapiro, N., Fundamentals of Engineering Thermodynamics, 4 <sup>th</sup> ed., John Wiley & Sons, 2010 6. Yunus A. Cengel, Michael A. Boles, Thermodynamics an engineering approach, 7 <sup>th</sup> ed., McGraw Hill, 2011
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
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2. Dr. S. Raja, National Aerospace Laboratories, Bangalore, raja@nal.res.in	2. Prof. Prasanta Kumar Das, IIT Kharagpur, pkd@mech.iitkgp.ernet.in.	2. Dr. T. Selvakumaran, SRMIST

Course Code	18ASC104J	Course Name	AIRCRAFT MATERIALS AND PRODUCTION TECHNIQUES	Course Category	C	Professional Core	L	T	P	C
							3	0	2	4

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1 :	Identify materials			
CLR-2 :	Utilize the mechanical behavior of materials			
CLR-3 :	Utilize the existing production technologies			
CLR-4 :	Identifying the selection of materials			
CLR-5 :	Identify material's Application			
CLR-6 :	Utilize the experience of machining Techniques for real-time applicaions			

Learning		
1	2	3
g (Bloom)	iciency (%)	ment (%)

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
nowledge	is	opment	n, Research	age	re	Sustainability			mm Work		inance	ing		

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLO-1 :	Identify materials and it properties	2	80	70	H	H	-	-	-	-	-	-	-	-	-	-	-	-	-
CLO-2 :	Analyze the application of materials in different aircraft components	2	85	75	H	H	-	-	H	-	-	-	-	-	-	-	-	-	-
CLO-3 :	Identify different treatments to strengthen materials	2	75	70	H	-	H	H	-	-	-	-	-	-	-	-	-	-	-
CLO-4 :	Identify different casting techniques	2	85	80	H	H	-	-	H	-	-	-	-	-	-	-	-	-	-
CLO-5 :	Analyze machining techniques	2	85	75	H	-	H	-	-	-	-	-	-	-	-	-	-	-	-
CLO-6 :	Analyze forming Techniques	2	80	70	-	-	-	-	-	-	-	-	-	-	-	-	-	-	-

Duration (hour)	15	15	15	15	15
S-1	SLO-1 Introduction to materials, mechanical properties	Heat Treatment	Casting Introduction	Mechanical working of Materials	Machining Process
	SLO-2 Fixed-wing aircraft structures	Purpose of Heat Treatment	Basic Terms	Introduction to mechanical Working	Introduction to Machines
S-2	SLO-1 Classification of aircraft materials	Principles of Heat Treatment	Casting Procedure	Hot Working	Lathe
	SLO-2 Materials used for aircraft components	Stages of Heat Treatment	Casting Nomenclature	Cold Working	Lathe Components, tools
S-3	SLO-1 Helicopter structures	Stages of Heat Treatment, Description	Sand Casting	Hot Working- Forging	Working of Lathe
	SLO-2 Space shuttle structures	Types of Heat treatment	Making of Sand Casting, Gating and risering System	Forging Types, Forging Defects	Operations in Lathe, tools
S 4-5	SLO-1 Lab 1: Step Turning	Lab 4: Drilling and Boring	Lab 7 Surface Grinding	Lab 10: Spur Gear Milling	Lab 13: Thread Cutting
	SLO-2				
S-6	SLO-1 Materials used in jet engines	Heat treatment of carbon steel	Special Casting Process	Rolling, Types of Rolling, Rolling Mills	Drilling Machine, Types of Drilling machine
	SLO-2 Light weight material for MAV/UAV.	Procedures of Heat treatment of carbon steel	Special casting process	Rolling Defects	Operations, Tools used in Drilling Machine
S-7	SLO-1 Super alloys.	Heat treatment of - aluminum alloys,	Expandable Mold Casting	Drawing	Shaper Machine
	SLO-2 Application of Composite materials	Procedures of Heat treatment of - aluminum alloys,	Shell Mold Casting	Drawing Types	Operations
S-8	SLO-1 Introduction to smart materials,	Heat treatment of titanium alloys.	Investment Casting	Extrusion	Quick return Mechanism



	SLO-2	Shape memory alloys	Procedures of Heat treatment of titanium alloys	Investment Casting Process	Extrusion Types	Mechanism Detail
S 9-10	SLO-1 SLO-2	Lab 2: Taper Turning	Lab 5: Shaper	Lab 8 Cylindrical Grinding	Lab 11: Helical Gear Milling	Lab 14: Slotting
S-11	SLO-1	Advanced structure ceramic	Heat treatment of Magnesium alloys.	Permanent Mold Casting, Die Casting, Centrifugal Casting, Casting Defects	Sheet Metal Operations, Shearing Operations	Slotter machine, mechanisms, Grinding Machines
	SLO-2	intermetallics, Ni and Ti aluminide	Procedures of Heat treatment of Magnesium alloys	Casting Defects	Types of Shearing Dies	Cutting Tools in Grinding Machines
S-12	SLO-1	Introduction to FRP,	Case Hardening	Welding Introduction	Forming Operations	Operations in Grinding Machines
	SLO-2	Glass and Carbon Composites	Procedures of Case Hardening	Gas Welding, Arc Welding	Forming Operations	Types of Grinding Machines
S-13	SLO-1	Aerospace Applications – Plastics and Rubber.	Stress relieving Procedures	Laser Beam Welding	Cutting Tools in sheet metal Process	Milling
	SLO-2	Emerging trends in Aerospace materials,	Protective Treatments	Electron Beam Welding, Electric Resistance Welding	Striking Tools in Sheet Metals, Riveting	Milling Operations, Types of Milling Machines
S 14-15	SLO-1 SLO-2	Lab 3: Taper boring	Lab 6: Drilling, Reaming & Tapping	Lab 9: Grooving and Knurling	Lab 12: External keyway cutting	Lab 15: Gear hobbing

Learning Resources	1. Adrian P. Mouritz, Introduction to aerospace materials, Woodhead Publishing Limited, 2012 2. Dieter, G. E., Mechanical Metallurgy, McGraw Hill, Singapore, 2001 3. Keshu S.C, Ganapathy K.K, Aircraft production technique, Interline Publishing House, Bangalore 1993 4. Dr. P C Sharma, A Text book of Production Technology, 8 <sup>th</sup> ed., S. CHAND and company Pvt. Ltd. 2014
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
	Understand										
Level 2	Apply	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
	Analyze										
Level 3	Evaluate	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. Vinay Kumar Gupta, National Physical Laboratory, guptavinay@nplindia.org	1. Dr. Srinivasa Rao Bakshi, IITM, Chennai, sbakshi@iitm.ac.in	1. Mr. N Bharat, SRMIST
2. Dr. S. Raja, National Aerospace Laboratories, Bangalore, raja@nal.res.in	2. Dr. Ramesh Babu, N , nrbabu@iitm.ac.in	2. Mr. K B Ravichandra kumar, SRMIST

Course Code	18ASC105T	Course Name	AIRCRAFT SYSTEMS AND INSTRUMENTS	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards	Nil		

Course Learning Rationale (CLR):		The purpose of learning this course is to:			Learning			Program Learning Outcomes (PLO)														
CLR-1 :	Identify the type of control system and its components used in aircraft.				Level of Thinking (Bloom)	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15		
CLR-2 :	Layout the components and accessories of hydraulic & Pneumatic system.																					
CLR-3 :	Identify the type of powerplant and various system operations in aircraft engines																					
CLR-4 :	Demonstrate the cabin environmental control system, oxygen system and other auxiliary system of an airplane.																					
CLR-5 :	Identify the various aircraft instruments and their functions.																					
CLR-6 :	Utilize the knowledge acquired for design, development & maintenance of aircraft & aero engine systems.																					
Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:																				
CLO-1 :	Understand the operation of various control system in an airplane				2	80	70															
CLO-2 :	Acquire knowledge on hydraulic and pneumatic system of modern jet airliner.				2	80	70															
CLO-3 :	Learn the working of various systems of piston and gas turbine engine				2	80	70															
CLO-4 :	Appreciate the need and functions of Cabin Environmental Systems and auxiliary systems of aircraft.				2	80	70															
CLO-5 :	Gain knowledge on principle and operation of various aircraft instruments.				2	80	70															
CLO-6 :	Acquire comprehensive knowledge of aircraft systems, engine systems and its instrumentation.				2	80	70															

Duration (hour)		9	9	9	9	9
S-1	SLO-1	Need for Control Systems	Introduction to Hydraulic Systems	Introduction to Aircraft Engines	Introduction to Cabin Environmental Control Systems	Introduction to Aircraft Flight Instruments
	SLO-2	Conventional Flight Controls.	Applications & Advantages	Types, Abnormal Combustion	Need for Aircraft Pressurization System	Types
S-2	SLO-1	Components of Conventional Flight Control System and their functions	Selection & Classification of Hydraulic Fluids	Introduction to Aircraft Fuel System	Principle of Air Cycle Cooling System	Principle of Air Data Instruments
	SLO-2	Push Pull rod System	Open Centre & Closed Centre System	Types of Fuel & Fuel System Components	Operation & Advantages	Operation of Altimeter
S-3	SLO-1	Cable Pulley System	Components of Hydraulic System and its functions	Gravity Feed Fuel System	Principle of Vapour Cycle Cooling System	Operation of Air Speed Indicator
	SLO-2	Disadvantages of Mechanical Control System.	Automatic Operating Control Valves	Pressure Feed Fuel System	Its Operation & Advantages	Operation of Vertical Speed Indicator
S-4	SLO-1	Challenges in Power Assisted Flight Control System	Study of Typical Hydraulic System for Modern Jet Airliner	Need for Lubrication System	Need for Cabin Heating System	Principle of Gyroscopic Instruments
	SLO-2	Q – Feel System	Operation and its Advantages	Functions and Characteristics of Lubricating Oil.	Types & Operation	Operation of Attitude Indicator
S-5	SLO-1	Servo Tabs	Aircraft Brake System	Types of Lubrication System, Wet Sump System	Need for Aircraft Oxygen System	Operation of Turn Coordinator
	SLO-2	Fully Powered Flight Control System for heavv aircraft	Types and Applications	Dry Sump System and their Advantages	Types & Advantages	Operation of Heading Indicator

S-6	SLO-1	Fly by Wire System (FBW)	Introduction to Pneumatic Systems	Need For Ignition System	Components of Oxygen System	Principle & Operation of Engine Instruments – Tachometer & EGT
	SLO-2	Operation of FBW & its Advantages	Applications & Advantages	Types of Ignition Systems	Its Operation	Principle & Operation of EPR, CHT & Manifold Pressure Gauge
S-7	SLO-1	Digital Fly by Wire System (DFBW)	Study of Typical Pneumatic System for Modern Airliner	Magneto Ignition System & its Operation	Introduction to Fire Detection Systems	Principle & Operation of Electronic Instruments – EADI & EHSI
	SLO-2	Operation of DFBW & its Advantages	Operation and its Advantages	Components of Ignition System of Gas Turbine Engine	Requirements for Fire Detection System	Principle & Operation of Electronic Systems Monitor Displays
S-8	SLO-1	Need for Automatic Flight Control Systems	Introduction to Landing Gear System	Need for Starting System	Types	Principle & Operation of EICAS
	SLO-2	Operation of Autopilot System	Classification of Landing Gear System	Types of starters	Principle and Operation	Need for Instrument Landing System (ILS)
S-9	SLO-1	Auto Throttle System (ATS)	Components of Landing Gear System	Pneumatic Starting System for Modern airliner	Need for Anti-Icing & De-Icing System	Components of ILS and their functions
	SLO-2	Advantages of ATS	Applications	Advantages of Pneumatic Starting System	Types and Applications.	Advantages

Learning Resources	<ol style="list-style-type: none"> <li>1. Ian Moir, Allan Seabridge, Aircraft Systems – Mechanical, Electrical and Avionics subsystems integration, 3<sup>rd</sup> ed., Professional Engineering Publishing Limited, 2008</li> <li>2. E.H.J.Pallet, Aircraft Instruments, 2<sup>nd</sup> ed., Pearson Publishing Company, 2009</li> <li>3. Aviation Maintenance Technician Handbook – Airframe, Vol.2, U.S.Dept. of Transportation, Federal Aviation Administration, Flight Standards Service, 2012</li> <li>4. Aviation Maintenance Technician Handbook – Powerplant, Vol.1, 2, U.S.Dept. of Transportation, Federal Aviation Administration, Flight Standards Service, 2012</li> <li>5. Michael J.Kroes, William A.Watkins ad Frank Delp, Aircraft Maintenance and Repair, 7<sup>th</sup> ed., Tata McGraw Hill, 2013</li> <li>6. Irwin Treager, Aircraft Gas Turbine Engine Technology, 3<sup>rd</sup> ed., McGraw-Hill, 1997</li> <li>7. The Jet Engine, 5<sup>th</sup> ed., Rolls Royce, Wiley Publication, 2005</li> </ol>
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com	1. Dr. A. P. Haran, Park College of Engineering & Technology, ap_haran@rediffmail.com	1. Dr. S. Sivakumar, SRMIST
2. Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in	2. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	2. Mr. G. Mahendra Perumal, SRMIST

Course Code	18ASC201J	Course Name	APPLIED SOLID MECHANICS	Course Category	C	Professional Core			
						L	T	P	C
						3	0	2	4

Pre-requisite Courses	18ASC101T	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):	The purpose of learning this course is to:	Learning			Program Learning Outcomes (PLO)														
CLR-1 :	Identify the stresses generated and structural changes in different bar materials subjected to different loads	1	2	3	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
CLR-2 :	Identify the variation of shear force, bending moments and bending stress in various beams subjected to different loads	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLR-3 :	Know the variation of curvature of beams subjected to loads based on which the slope, deflection calculations be made																		
CLR-4 :	Identify the advantages and disadvantages of using solid and hallow shafts, different springs for different loads																		
CLR-5 :	Know the buckling characteristics of column for various end conditions and stresses generated in thin and thick cylinders																		
CLR-6 :	Know the behavior of different structural materials for different types of loading																		

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Learning			Program Learning Outcomes (PLO)														
CLO-1 :	Differentiate a ductile material and a brittle material after performing a tension test	2	80	70	H	H	-	-	-	-	-	-	-	-	-	-	-	-	-
CLO-2 :	Analyze the shear force and bending diagrams in cantilever and simply supported beams	2	80	75	H	H	-	H	-	-	-	-	-	-	-	-	-	-	-
CLO-3 :	Make calculations for the design of a beam based on the bending stress and desired deflection	2	75	70	H	H	H	H	-	-	-	-	-	-	-	-	-	-	-
CLO-4 :	Design the shaft for a particular torque transmission and springs for energy absorption	2	80	75	H	H	H	H	-	-	-	-	-	-	-	-	-	-	-
CLO-5 :	Find the planes of principal stresses in a stressed model and hoop stress, longitudinal stress in thin walled pressure vessel	2	85	75	H	-	H	-	-	-	-	-	-	-	-	-	-	-	-
CLO-6 :	Calculate the various stresses generated in a particular element subjected to different loading	2	80	70	H	H	H	H	-	-	-	-	-	-	-	-	-	-	-

Duration (hour)		15	15	15	15	15
S-1	SLO-1	Concept of stress and strain in a bar element	Hardness and Strength	Relation between deflection, slope, radius of curvature	Theory of pure torsion	Stresses on inclined planes
	SLO-2	Hooke's law, Poisson's ratio, Elastic young's modulus	Ductility and brittleness	Shear force and bending moment Derivation explanation for different loads	Explain shear stress variation in a circular (solid and hollow)	Derivation explanation
S-2	SLO-1	True and Engineering stress strain curve for ductile material in tension, compression	Difference between static loading and dynamic loading	Find slope and deflection in a cantilever beam by double integration method	Apply torsion equation based on allowable shear stress	Mohr's circle derivation
	SLO-2	True and Engineering stress strain curve for a brittle material in tension	Impact loading	Problem solving	Apply torsion equation based on allowable angle of twist	Plane stress case
S-3	SLO-1	Concept of shear stress, shear strain and Rigidity modulus	Statically Determinate structure, examples	Find slope and deflection in a simply supported beam by double integration	Compare solid and hollow shafts for transmission of same torque	Mohr's circle construction
	SLO-2	Principle of complementary shear	Statically Indeterminate structure, examples	Problem solving	Applications explanation	Procedure to different kinds of load
S-4-5	SLO-1	Lab1: Tension test	Lab-4: Brinell Hardness Test and Vickers hardness test	Lab-7: Deflection test in a cantilever beam with a point loads	Lab-10: Torsion test on a circular rod using digital torsion testing machine	Lab 13: Charpy Impact test and Izod impact test
	SLO-2					
S-6	SLO-1	Biaxial and triaxial state of stress and volumetric strain	Beam, types of beams, types of load	Find slope and deflection in a simply supported beam by Macaulay's method	Explain shear stress variation in closed coil helical sprigs	Concept of pure shear, relation between Young's, Shear and bulk modulus
	SLO-2	Problem solving	Procedure of solving a beam	Problem solving	Applications	Derivation explanation
S-7	SLO-1	Analysis of prismatic bar subjected to	Shear force and bending moment diagram	Moment Area Theorem-I	Explain shear stress variation in open coil	Numerical solving



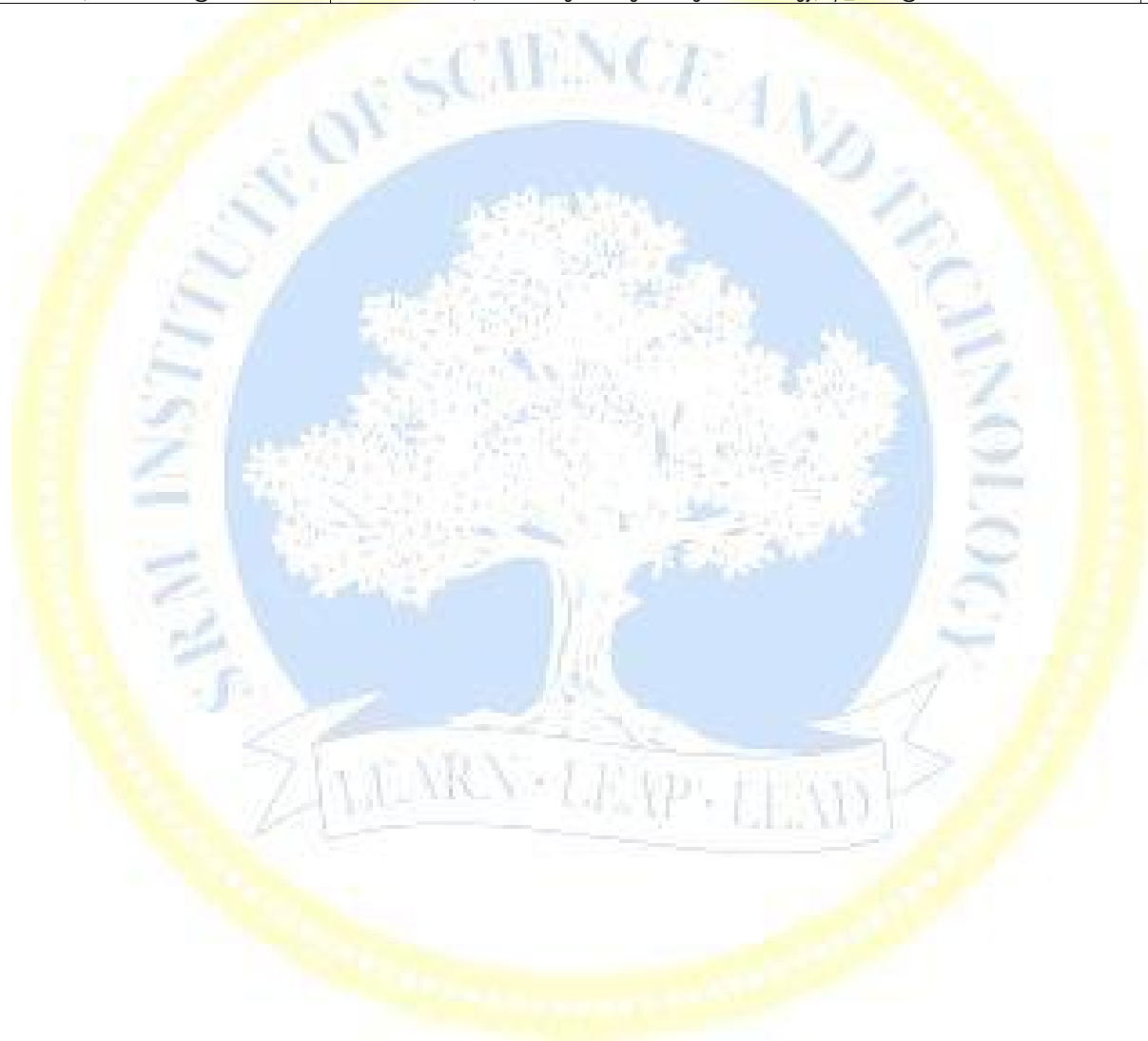
		single load	and their sign convention		helical springs	
	SLO-2	Analysis of prismatic bar subjected to varying loads	Shear force and bending moment diagram for a cantilever beam subjected to point load and UDL	Application to cantilever and simply supported beam	Numerical explanation	Numerical solving
S-8	SLO-1	Analysis of non-prismatic bar subjected to single load	Shear force and bending moment diagram for a simply supported beam subjected to point load	Moment Area Theorem-II	Stiffness of closed coil helical spring	Fatigue load-Explanation
	SLO-2	Analysis of non-prismatic bar subjected to varying loads	Shear force, bending moment diagram for a simply supported beam subjected to UDL	Application to cantilever and simply supported beams	Stiffness of open coil helical spring	S-N curve for various materials
S-9-10	SLO-1	Lab-2: Compression test	Lab-5: Repeat	Lab-8: Deflection test: Simply supported beam	Lab-11: Tension test on a closed coil helical spring	Lab-14: Determine endurance limit of the given material by performing a fatigue test.
	SLO-2					
S-11	SLO-1	Analysis of composite bars	Shear force and bending moment diagram for a overhanging beam	Principle of superposition	Shaft subjected to combined bending and torsion	Thin walled pressure vessel subjected to internal pressure
	SLO-2	Numerical solving	Point of contraflexure	Application	Derivation	Hoop stress explanation
S-12	SLO-1	Thermal stresses-Support Yield	Explanation of bending stress variation in a beam subjected to pure bending	Maxwell reciprocal theorem	Strain energy due bending	Thin walled cylindrical pressure vessel
	SLO-2	Numerical Solving	Application	Application	Derivation	Longitudinal stress explanation
S-13	SLO-1	Thermal stresses-composite bars	Explain shear stress variation in a beam of symmetrical and unsymmetrical cross sections subjected to bending	Explain shear force diagram of an aircraft wing	Strain energy due torsion	Thick cylinder
	SLO-2	Problem solving	Application	Explain bending moment diagram of an aircraft wing	Derivation	Lame's theory
S-14-15	SLO-1	Lab-3: Study of magnified images obtained using Inverted Metallurgical Microscope on a specimen.	Lab-6: Rockwell Hardness Test	Lab-9:Repeat	Lab-12: Compression test on an open coil helical spring	Lab-15 :Repeat
	SLO-2					

Learning Resources	1. Ferdinand P.Beer, Russell Johnston, John T.Dewolf, Mechanics of Materials, SI Metric, 3 <sup>rd</sup> ed., Tata McGraw-Hill Education, 2011 2. Egor P. Popov., Engineering Mechanics of Solids, 2 <sup>nd</sup> ed., Prentice Hall of India, 2009	3. James M. Gere, Mechanics of Materials, 8 <sup>th</sup> ed., Brooks/Cole, USA, 2013 4. Shigley, J. E., Applied Mechanics of Materials, International Student Edition, McGraw Hill, 2000 5. V. Feodosyev. Strength of Materials, MIR Publishers, Moscow 1968
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember Understand	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
Level 2	Apply Analyze	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
Level 3	Evaluate Create	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in	1. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	1. Mr. S. Chandra Sekhar, SRMIST
2. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com	2. Dr. A. P. Haran, Park College of Engineering & Technology, ap_haran@rediffmail.com	2. Mr. K B Ravichandra kumar, SRMIST



Course Code	18ASC202J	Course Name	INCOMPRESSIBLE AERODYNAMICS	Course Category	C	Professional Core			
						L	T	P	C
						3	0	2	4

Pre-requisite Courses	18ASC102J	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):		The purpose of learning this course is to:			Learning			Program Learning Outcomes (PLO)																
CLR-1 :	Identify and utilize the lift generating devices				Level of Thinking (Bloom)	1	2	3	Engineering Knowledge	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
CLR-2 :	Evaluate the forces and moments acting on aero foils and wings under ideal flow conditions.																							
CLR-3 :	Evaluate and optimize the aerofoil characteristics																							
CLR-4 :	Evaluate and optimize the wing characteristics.																							
CLR-5 :	Evaluate and optimize the propeller characteristics.																							
CLR-6 :	Evaluate and optimize the aerodynamic interaction effects between different components of aircraft																							
Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:																						
CLO-1 :	Understand the lift generation and lift generating devices					1	80	75	M	M	M	H	M	-	-	-	-	-	-	-	M	-	H	-
CLO-2 :	Analyze the forces and moments acting on aero foils and wings under ideal flow conditions.					2	80	75	H	H	H	H	H	-	-	-	-	-	-	-	H	-	H	-
CLO-3 :	Analyze the aerofoil characteristics.					3	70	60	H	H	H	H	H	-	-	-	-	-	-	-	H	-	H	-
CLO-4 :	Analyze the wing characteristics.					3	70	60	H	H	H	H	H	-	-	-	-	-	-	-	H	-	H	-
CLO-5 :	Analyze the propeller characteristics.					3	70	60	H	H	H	H	H	-	-	-	-	-	-	-	H	-	H	-
CLO-6 :	Analyze the aerodynamic interaction effects between different components of aircraft					2	70	65	H	H	H	H	H	-	-	-	-	-	-	-	H	-	H	-

Duration (hour)		15	15	15	15	15
S-1	SLO-1	Introduction to aerodynamics	Center of pressure	High lift devices	torque grading and efficiency equation	Influence of taper and twist applied to wings
	SLO-2	Introduction to the mathematical model of flow	Aerodynamic center	Thin airfoil theory-Flapped airfoil - circulation equation	Combined blade element and momentum theories velocity comparison	effect of sweep back and delta wings
S-2	SLO-1	Airfoil geometry	Numerical problems on Center of pressure	Thin airfoil theory-Flapped airfoil - coefficient of lift and moment	Comparison of thrust and torque equation	Fundamentals of potential flows
	SLO-2	Airfoil nomenclature	Numerical problems on Aerodynamic center	Geometry of the propeller	Axial flow factor equation	Indirect and direct methods of mathematical model of flow
S-3	SLO-1	Wing geometry parameters	Numerical problems on Center of pressure and Aerodynamic center	Forces acting on Propeller	Angular flow factor equation	Basic equations of potential flow
	SLO-2	Application of wing geometry parameters	Experimental characteristics of airfoil	Types of Propeller	The Biot-savart law	Introduction to panel methods
S-4-5	SLO-1	Lab 1: Introduction to subsonic wind tunnel	Lab 4: Study of flow over streamlined body by laser beam assisted smoke visualization technique	Lab 7: Study of flow over a tapered finite wing with wingtip by laser beam assisted flow visualization technique	Lab 10: Pressure distribution and Estimation of forces acting over a rough cylinder	Lab 13: Estimation of forces acting over a streamlined body using force balance method
	SLO-2					
S-6	SLO-1	Vortex motions, vortex filament, vortex sheet	Thin airfoil theory assumptions and limitations.	Propeller arrangements	Application of Biot-savart law	Source panel method
	SLO-2	Vortex types, Starting vortex, trailing edge vortex	Fundamental equation of thin airfoil theory	Axial momentum theory assumptions and limitations.	Introduction of Prandtl's lifting line theory	Application of Source panel method

S-7	SLO-1	kutta's and kelvins theorem	Thin airfoil theory- symmetrical airfoil – circulation equation	Pressure and velocity distribution across the propeller control volume	Governing equations of Prandtl's lifting line theory	Vortex panel method
	SLO-2	Kutta – Joukowski theorem	Thin airfoil theory- symmetrical airfoil – coefficient of lift and moment	Propeller thrust equation	Applications of Prandtl's lifting line theory	Application of Vortex panel method
S-8	SLO-1	Lift generation	Thin airfoil theory- symmetrical airfoil – location of forces	Propeller power equation	Prandtl theory- Elliptical lift Distribution – circulation equation	Wing- fuselage interference
	SLO-2	bound and horseshoe vortex	Numerical problems on thin airfoil theory	efficiency equation and comparison	Prandtl theory- Elliptical lift Distribution – downwash and induced angle	Wing-engine interference
S 9-10	SLO-1	Lab 2: Wind tunnel measurement techniques	Lab 5: Study of Magnus effect using rotating cylinder by laser beam assisted smoke visualization technique.	Lab 8: Calibration of subsonic wind tunnel	Lab 11: Pressure distribution and Estimation of forces acting over a sphere model	Lab 14: Estimation of pressure distribution acting over a symmetrical / unsymmetrical airfoil for different angle of attack
	SLO-2					
S-11	SLO-1	Aerodynamic forces	Thin airfoil theory- unsymmetrical airfoil- circulation equation	Numerical problems on Axial momentum theory	Prandtl theory- Elliptical lift Distribution- coefficient of lift and induced drag	Wing-landing gear interference
	SLO-2	Aerodynamic moments	Thin airfoil theory- unsymmetrical airfoil – verification of circulation equation	Numerical problems on Axial momentum theory	Fundamentals of Prandtl theory- General lift Distribution	Wing – propeller interference
S-12	SLO-1	Types of drag	Thin airfoil theory- unsymmetrical airfoil – coefficient of lift	Numerical problems on Axial momentum theory	Prandtl theory- General lift Distribution- circulation equation	Wing –tail interference
	SLO-2	Numerical problems on Aerodynamic forces	Thin airfoil theory- unsymmetrical airfoil – coefficient of moment and location of forces	fundamentals of blade-element theory	Prandtl theory- General lift Distribution- coefficient of lift and induced drag	interference flow over an airplane as a whole
S-13	SLO-1	Numerical problems on Aerodynamic forces and moments	Numerical problems on thin airfoil theory	Basic equation of thrust and torque grading	Lift slope relation	Passive Laminar flow control methods
	SLO-2	Numerical problems on Aerodynamic forces and moments	Numerical problems on thin airfoil theory	thrust grading equation	Numerical problems on Lift slope relation	Active Laminar flow control methods
S 14-15	SLO-1	Lab 3: Study of flow over bluff body by laser beam assisted smoke visualization technique	Lab 6: Study of flow over a tapered finite wing without wingtip by laser beam assisted flow visualization technique	Lab 9: Pressure distribution and Estimation of forces acting over a smooth cylinder	Lab 12: Estimation of forces acting over a bluff body using force balance method	Lab 15: Estimation of forces acting over a symmetrical / unsymmetrical airfoil for different angle of attack
	SLO-2					

Learning Resources	1. Houghton, E, L., Carruthers, N, B., Aerodynamics for Engineering Students, 6 <sup>th</sup> ed., Edward Arnold Publishers Ltd., London, 2012	3. Clancy, L, J., Aerodynamics, Pitman, 1986
	2. Anderson, J,D., Fundamentals of Aerodynamics, 6 <sup>th</sup> ed., McGraw Hill, 2016	4. Milne, L.H., Thomson, Theoretical Aerodynamics, Dover, 1985

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember Understand	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
Level 2	Apply Analyze	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
Level 3	Evaluate Create	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,



Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in	1. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	1. Mr. R. Mohamed Arif, SRMIST
2. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com	2. Dr. P. K Dash, Nitte Meenakshi Institute of Technology, Bangalore, drpdash@gmail.com	2. Mr. K B Ravichandra kumar, SRMIST



Course Code	18ASC203T	Course Name	AIR BREATHING PROPULSION	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering			Data Book / Codes/Standards	Nil

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1 :	Identify the working principles of gas turbine propulsion systems			
CLR-2 :	Design of inlets, combustion chambers, nozzles used in Air breathing engines			
CLR-3 :	Design of compressors in gas turbine propulsion systems			
CLR-4 :	Design of turbines in gas turbine propulsion systems			
CLR-5 :	Understand the principle of operation of Pulse jet, RAMJET and SCRAMJET engines			
CLR-6 :	Understand the working principles of gas turbine propulsion systems			

Learning			
1	2	3	
Thinking (Bloom)	Efficiency (%)	Attainment (%)	

Program Learning Outcomes (PLO)																			
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15					
Knowledge	Analysis	Development	Design, Research	Usage	Culture	Sustainability													

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLO-1 :	Analyze the performance and component efficiencies of gas turbine propulsion systems	2	80	70	H	-	-	-	-	-	-	-	-	-	-	-	-	-	-
CLO-2 :	Analyze inlets, combustion chambers, nozzles used in Air breathing engines	2	85	75	H	H	H	-	-	-	-	-	-	-	-	-	-	-	-
CLO-3 :	Analyze the compressors in gas turbine propulsion systems	2	75	70	H	-	-	H	-	-	-	-	-	-	-	-	-	-	-
CLO-4 :	Analyze the turbines in gas turbine propulsion systems	2	85	80	H	H	-	-	-	-	-	-	-	-	-	-	-	-	-
CLO-5 :	Analyze the performance of Pulse Jet, RAMJET and SCRAMJET engines	2	85	75	H	-	H	-	-	-	-	-	-	-	-	-	-	-	-
CLO-6 :	Analyze the performance and component efficiencies of gas turbine propulsion systems	2	80	70	H	-	-	-	-	-	-	-	-	-	-	-	-	-	-

Duration (hour)	9	9	9	9	9
S-1	SLO-1 Introduction to Air breathing engines	Inlets	Compressor	Turbine	Pulse Jet Engine-Operating Principle
	SLO-2 Ideal and actual Brayton cycle	Classification of Inlets	Classification of compressors	Classification of turbines	RAMJET Engine
S-2	SLO-1 Turbojet Engine	Subsonic Inlets	Axial flow compressor	Axial flow turbine stage	Working of RAMJET
	SLO-2 Numerical Problems on turbojet engine	Supersonic Inlets	Work and compression ratio	Velocity triangles and Power output	Working of RAMJET
S-3	SLO-1 High bypass turbofan Engine	Modes of Inlet operation	Degree of reaction	Blade Element theory	RAMJET with afterburner
	SLO-2 Low bypass turbofan engine	Starting problems and Shock swallowing methods	Characteristic performance of a single stage axial compressor	Blade Element theory	RAMJET with afterburner
S-4	SLO-1 Numerical Problems on turbofan engine	Numerical Problems on Inlets	Characteristic performance of a multistage axial compressor	Free vortex theory	RAMJET performance
	SLO-2 Numerical Problems on turbofan engine	Numerical Problems on Inlets	Cascading of axial compressor-Compressor efficiency	Free vortex theory	RAMJET performance
S-5	SLO-1 Turboshaft engine	Gas turbine combustion chamber	Numerical Problems on Single stage Axial flow compressor	Limiting Factors of gas turbine design	Numerical Problems on RAMJET
	SLO-2 Turboprop engine	Types of combustion chamber	Numerical Problems on Single stage Axial flow compressor	Limiting Factors of gas turbine design	Numerical Problems on RAMJET
S-6	SLO-1 Numerical Problems on turboprop engine	Fuel injector- Flame Tube cooling	Numerical Problems on multi stage Axial flow compressor	Turbine performance	SCRAMJET Engine
	SLO-2 Numerical Problems on turboprop engine	Flame Stabilization-Flame holders	Numerical Problems on multi stage Axial	Turbine blade cooling	Working principle of SCRAMJET Engine

				flow compressor		
S-7	SLO-1	Typical engine performance	Nozzle	Centrifugal compressor	Turbine blade cooling methods	Problems faced in supersonic combustion
	SLO-2	Typical engine performance	Classification of nozzles	Working Principle of a centrifugal compressor	Turbine and compressor matching	Problems faced in supersonic combustion
S-8	SLO-1	Methods of thrust augmentation	Numerical Problems on Nozzles.	Work and compression ratio	Numerical Problems on Axial flow turbine	Numerical Problems on SCRAMJET
	SLO-2	Methods of thrust augmentation	Numerical Problems on Nozzles.	Work and compression ratio	Numerical Problems on Axial flow turbine	Numerical Problems on SCRAMJET
S-9	SLO-1	Introduction to Air breathing engines	Inlets	Compressor	Turbine	Pulse Jet Engine-Operating Principle
	SLO-2	Ideal and actual Brayton cycle	Classification of Inlets	Classification of compressors	Classification of turbines	RAMJET Engine

Learning Resources	1. Hill, P. G., Peterson, C. R., <i>Mechanics and Thermodynamics of Propulsion</i> , 2 <sup>nd</sup> ed., Addison-Wesley Publishing Company, 1992. 2. Cohen, H. Rogers. G.F.C., Saravanamuttoo. H.I.H., <i>Gas turbine theory</i> . 4 <sup>th</sup> ed., Pearson education 3. V.Ganesan., <i>Gas Turbines</i> , 3 <sup>rd</sup> ed., Tata McGraw-Hill Education, 2010	4. Rolls-Royce , <i>Jet Engine Manual</i> , 3rd edition, 1983 5. Oats, G.C., <i>Aerothermodynamics of Aircraft Engine Components</i> , AIAA Education Series, 1985 6. Mattingly, J.D., Heiser, W.H., Pratt, D.T., <i>Aircraft Engine Design</i> , AIAA Education Series, 2002
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#### Learning Assessment

	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember Understand	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
Level 2	Apply Analyze	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
Level 3	Evaluate Create	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

#### Course Designers

Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in	1. Dr. P. K Dash, Nitte Meenakshi Institute of Technology, Bangalore, drpdash@gmail.com	1. Mr. G. Saravanan, SRMIST
2. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com	2. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	2. Mr. G. Mahendra Perumal, SRMIST

# **ACADEMIC CURRICULA**

## **Professional Core Courses**

### **AEROSPACE ENGINEERING**

**Regulations - 2018**



**SRM INSTITUTE OF SCIENCE AND TECHNOLOGY**

**(Deemed to be University u/s 3 of UGC Act, 1956)**

**Kattankulathur, Kancheepuram, Tamil Nadu, India**



Course Code	18ASC301J	Course Name	COMPRESSIBLE AERODYNAMICS	Course Category	C	Professional Core	L	T	P	C
							3	0	2	4

Pre-requisite Courses	18ASC202J	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards	Gas tables		

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1:	Identify the different wave types and wave propagation			
CLR-2:	Evaluate the change in properties across shock waves and optimize the supersonic vehicle design			
CLR-3:	Evaluate the change in properties across expansion waves			
CLR-4:	Evaluate and optimize the nozzle flow characteristics.			
CLR-5:	Design and Evaluate the duct flow with heat transfer and frictional effect.			
CLR-6:	Design and optimize the supersonic nozzle			

Learning				Program Learning Outcomes (PLO)														
1	2	3		1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Thinking (Bloom)	Efficiency (%)	Attainment (%)		Knowledge	Analysis	Development	Design, Research	Usage	Culture	Sustainability		Team Work	Communication	Finance & Economics	Learning			

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLO-1:	Understand the properties of different waves in supersonic flow	1	90	80	H	M	M	M	M	-	-	-	-	-	-	M	-	H	-
CLO-2:	Analyze the properties of shock wave	2	80	75	H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
CLO-3:	Analyze the properties of expansion wave	3	80	75	H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
CLO-4:	Analyze the flow through nozzle	3	70	60	H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
CLO-5:	Analyze the heat transfer and frictional effect in flow	3	80	75	H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
CLO-6:	Analyze the compressibility effects and design the supersonic nozzle	2	70	60	H	H	H	H	H	-	-	-	-	-	-	H	-	H	-

Duration (hour)		15	15	15	15	15
S-1	SLO-1	Introduction to compressible flow	Normal shock wave properties – Mach number relation	Multiple shock system	Numerical problems on Nozzle flow relations	Fanno curve
	SLO-2	Basic thermodynamic concepts	Normal shock wave properties – density relation	Numerical problems on multiple shock systems	Numerical problems on Nozzle flow relations	Numerical problems on Fanno flow
S-2	SLO-1	Basic thermodynamic equations	Normal shock wave properties – pressure relation	The propagating shock wave	Numerical problems on Nozzle flow relations	Numerical problems on Fanno flow
	SLO-2	Momentum and Energy equations for compressible fluid flow	Normal shock wave properties – temperature relation	Numerical problems on propagating shock wave	Rayleigh flow equations – static properties	Numerical problems on Fanno flow
S-3	SLO-1	Wave propagation	Normal shock wave properties – entropy change	Governing equation of Prandtl-meyer expansion waves	Rayleigh flow equations – stagnation properties	Velocity potential equation for compressible flow
	SLO-2	Shock formation	Hugoniot equation	Expression of Prandtl –meyer function	variation of flow properties for subsonic flow	Small perturbation theory
S 4-5	SLO-1	Lab 1: Study of various types of Supersonic wind tunnel.	Lab 4: Calibration of supersonic wind tunnel	Lab 7: Visualization of shock wave pattern on ramp model using Schlieren flow visualization technique	Lab 10: Visualization of shock wave pattern on Diamond Airfoil using Schlieren flow visualization technique	Lab 13: Investigation of intersection of right and left running shock waves for various deflection angles using Schlieren flow visualization
	SLO-2					
S-6	SLO-1	types of waves	Numerical problems on normal shock	Numerical problems on expansion waves	variation of flow properties for supersonic flow	Small perturbation theory
	SLO-2	Speed of sound derivation	Numerical problems on normal shock	Numerical problems on expansion waves	Critical reference states	Linearized pressure coefficient

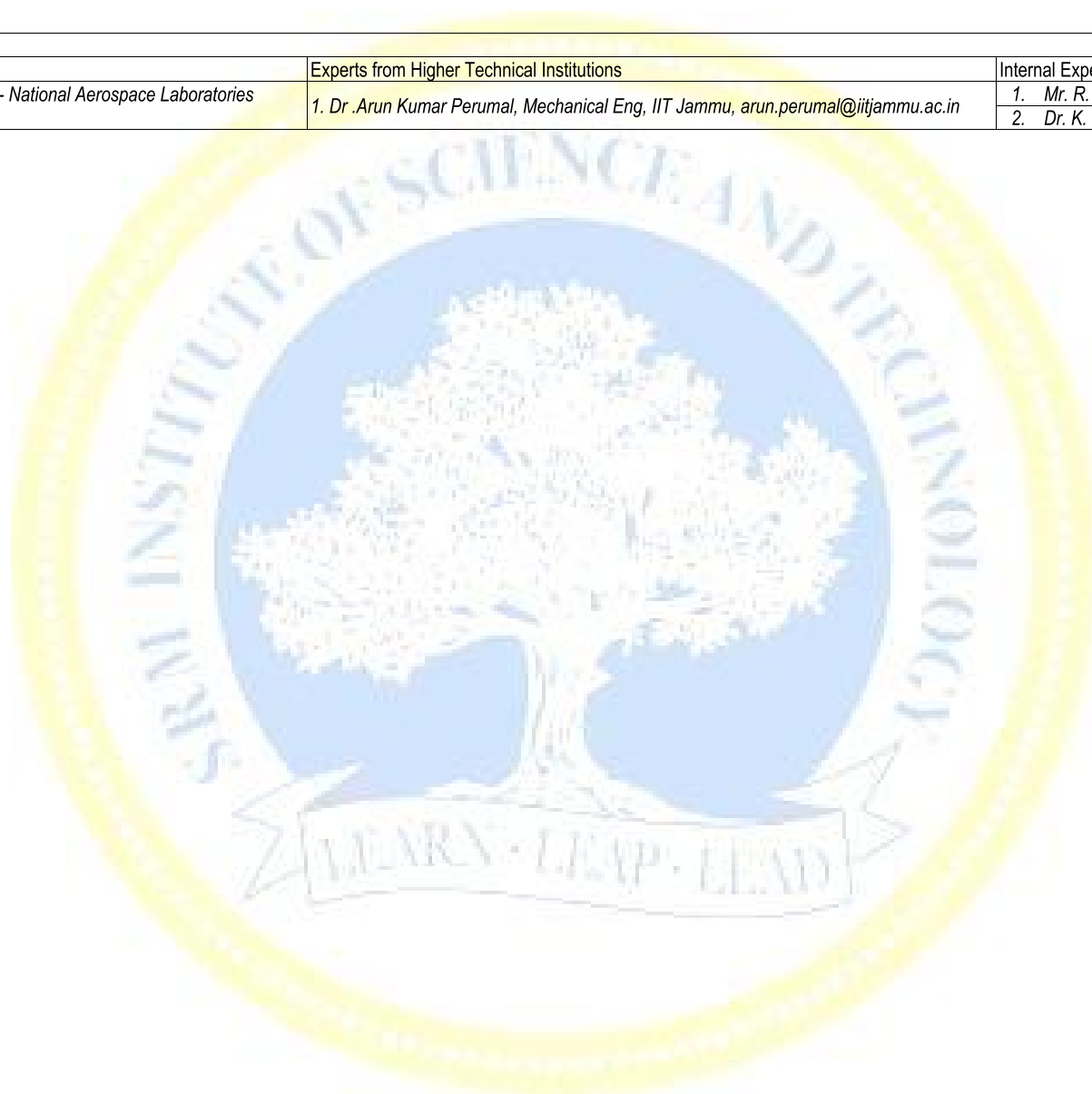
Duration (hour)		15	15	15	15	15
S-7	SLO-1	Change in entropy relation	Numerical problems on normal shock	Shock Expansion theory – flat plate	Rayleigh curve	Linearized pressure coefficient
	SLO-2	Numerical problems on thermodynamic concepts	Oblique shock properties	Numerical problems on shock expansion theory	Numerical problems on Rayleigh flow	Prandtl-Glauert compressibility correction
S-8	SLO-1	Isentropic relations	$\Theta$ - $\beta$ -M relation and graph	Shock Expansion theory – Diamond airfoil	Numerical problems on Rayleigh flow	Supersonic linearized theory
	SLO-2	Numerical problems on isentropic relations	Supersonic flow over wedges and cones	Numerical problems on shock expansion theory	Numerical problems on Rayleigh flow	Application of Supersonic linearized theory
S-9-10	SLO-1	Lab 2: Study of various Supersonic flow visualization techniques and its applications.	Lab 5: Mach number distribution of different area ratio C-D nozzles	Lab 8: Visualization of shock wave pattern on wedge model using Schlieren flow visualization technique	Lab 11: Investigation of supersonic flow over different aircraft/ missile models using Schlieren flow visualization technique	Lab 14: Experimental study of supersonic jet
	SLO-2					
S-11	SLO-1	Isentropic relations – stagnation state	Shock polar	Nozzle flow relations: Area – velocity	Fanno flow- equations	Numerical problems on linearized theory
	SLO-2	Numerical problems on isentropic relations	Shock reflections	Supersonic nozzle and diffuser	Fanno flow- equations	Numerical problems on linearized theory
S-12	SLO-1	Isentropic relations – critical state	Shock interactions	Nozzle flow relations: Area – Mach number	variation of flow properties with duct length	Introduction to Method of characteristics
	SLO-2	Numerical problems on isentropic relations	Numerical problems on oblique shock	Nozzle flow relations: Maximum mass flow rate	variation of flow properties for subsonic flow	Method of characteristics
S-13	SLO-1	Characteristic Mach number	Numerical problems on oblique shock	Variation of Pressure and Mach number along the C-D nozzle	variation of flow properties for supersonic flow	Application of Method of characteristics
	SLO-2	Numerical problems on isentropic relations	Numerical problems on oblique shock reflections	Under expansion and over expansion nozzle	Critical reference states	Application of Method of characteristics
S-14-15	SLO-1	Lab 3: Study of SRM supersonic wind tunnel and its instrumentations.	Lab 6: Investigation of starting normal shock wave movement inside Convergent Divergent Nozzle.	Lab 9: Verification of “Three dimensional relieving effect”.	Lab 12: Investigation of detached shock wave pattern using Schlieren flow visualization technique	Lab 15: Noise measurement in supersonic jet
	SLO-2					

Learning Resources	1. Rathakrishnan, E., “Gas Dynamics”, Prentice Hall India Learning Private Limited, 6th edition, Delhi, India, 2016.	3. Shapiro, A.H., “The Dynamics and Thermodynamics of Compressible Fluid Flow (Vol I and Vol II)”, Ronald Press, 1953.
	2. Anderson J. D., Jr., “Modern Compressible Flow with Historical Perspective,” McGraw Hill Publishing Co., 3rd edition, 2017	4. Zucker, R. D., Biblarz, O., “Fundamentals of Gas Dynamics”, Wiley-Blackwell; Third edition ( 2019) 5. Yahya, S. M., “Fundamentals of Compressible Flow with Aircraft and Rocket Propulsion”, New Age International Publishers; Sixth edition (2018)

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
	Understand										
Level 2	Apply	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
	Analyze										
Level 3	Evaluate	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
<i>Dr Manishankar C, Senior Scientist, CSIR - National Aerospace Laboratories Bangalore.</i>	1. <i>Dr .Arun Kumar Perumal, Mechanical Eng, IIT Jammu, <a href="mailto:arun.perumal@iitjammu.ac.in">arun.perumal@iitjammu.ac.in</a></i>	1. <i>Mr. R. Mohamed Arif, SRMIST.</i> 2. <i>Dr. K. K. Bharadwaj, SRMIST.</i>



Course Code	18ASC302T	Course Name	FLIGHT DYNAMICS - I	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	18ASC202J	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards			

Course Learning Rationale (CLR):	The purpose of learning this course is to:	Learning			Program Learning Outcomes (PLO)														
CLR-1 :	Understand the art of application of aerodynamics knowledge into an aircraft	1	2	3	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
CLR-2 :	Know the performance of different powerplants at different flight conditions	Thinking (Bloom)	Proficiency (%)	Attainment (%)	Engineering Knowledge	Analysis	Development	Design,	Tool Usage	Culture	Environment & Sustainability	Team & Team Work	Communication	Project & Finance	Learning				
CLR-3 :	Know the various performance parameters of an airplane																		
CLR-4 :	Learn optimizing different performance parameters for maximizing range, endurance and fuel efficiency																		
CLR-5 :	Learn role of performance parameters in various maneuvers of the airplane																		
CLR-6 :	Learn experimental determination of various performance parameters of a real airplane by flight testing																		

Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:																		
CLO-1:	Estimate and optimize drag polar of an aircraft	3	85	90	H	M	M	H	L	-	H	-	-	-	-	H	H	H	-
CLO-2:	Select optimal powerplant for the required airplane with the desired performance	3	80	85	M	M	M	M	L	-	H	-	M	-	-	H	H	H	-
CLO-3:	Indicate the different flight techniques for optimized steady and accelerated flight performance	3	75	80	M	H	M	H	M	-	H	-	-	-	-	H	H	H	-
CLO-4:	Perform basic calibration of flight instrument data	3	50	60	M	M	M	M	L	-	L	-	M	-	-	H	H	H	M
CLO-5:	Perform conceptual design of any type of airplane	3	70	70	M	M	H	H	M	M	H	M	H	-	M	H	H	H	-
CLO-6:	Determine the drag polar of a real airplane with real flight testing	3	00	80	M	H	M	M	M	-	-	-	M	M	-	H	-	H	M

Duration (hour)		9	9	9	9	9
S-1	SLO-1	Forces and moments acting on a vehicle in flight	Thrust and propulsive efficiency of air breathing engine	Equation of motion for steady level flight	Range -Definition and conditions for maximum range for jet driven aircraft	Flight instruments- Air Data systems
	SLO-2	Coordinate systems	Thrust and propulsive efficiency of rocket engine	Conditions for maximum velocity in steady level flight	Range -Definition and conditions for maximum range for propeller driven aircraft	Altitude and speed definitions
S-2	SLO-1	Equations of motion of a rigid flight vehicle	Trade-off between thrust and propulsive efficiency	Conditions for minimum in steady level flight	Endurance – Definition and conditions for maximum endurance for jet driven airplane	Air speed indicator- construction and working
	SLO-2	Solving Problems	Solving Problems	Solving problems	Endurance – Definition and conditions for maximum endurance for propeller driven airplane	Air speed indicator- speed calibration
S-3	SLO-1	International Standard Atmosphere and various layers in ISA	TSFC and BSFC	Solving problems	Solving problems	Solving air speed calibration problems
	SLO-2	Derivation of hydrostatic equation	Variation of Thrust/Power and SFC with respect to velocity and altitude for reciprocating engine	Power required vs velocity – Graphical approach and analytical approach	Solving problems	Solving air speed calibration problems
S-4	SLO-1	Derivation of Pressure, Temperature and density in troposphere	The propeller – variation of propulsive efficiency with advance ratio	Power available and maximum velocity for propeller driven and jet driven aircraft	Accelerated flight- Level turn- Equations of motion	Altimeter- construction and working
	SLO-2	Solving Problems in Gradient layer	Constant speed propellers	Minimum velocity- Stall and high lift devices	Level turn- conditions for turn radius and turn rate	Altimeter- altitude corrections



Duration (hour)		9	9	9	9	9
S-5	SLO-1	Derivation of Pressure, Temperature and density in Stratosphere	Variation of Thrust/Power and SFC with respect to velocity and altitude for turbojet engine	Aerodynamic relations associated with maximum efficiency conditions	Level turn- constraints on load factor, Constraints on velocity,	Solving problems- altimeter
	SLO-2	Solving Problems in Iso thermal layer	Solving Problems	Solving problems	Level turn- calculation of minimum turn radius and minimum turn rate.	Rate of climb indicator – construction and working
S-6	SLO-1	Various types of drags acting on an airplane	Variation of Thrust/Power and SFC with respect to velocity and altitude for turbofan engine	Rate of climb- equation of motion	Solving problems	Mach meter- construction and working
	SLO-2	Methods to minimize various drags	Solving Problems	Rate of climb- Graphical approach and hodograph	Solving problems	Angle of attack indicator- construction and working
S-7	SLO-1	Interference Drag and methods to minimize it Drag polar of vehicles in subsonic speed	Variation of Thrust/Power and SFC with respect to velocity and altitude for turboprop engine	Analytical approach for maximum climb angle	Pull-up and pull-down maneuvers	Flight determination of drag polar- introduction
	SLO-2	Transonic Drag Divergence	Solving Problems	Analytical approach for maximum rate of climb	V-n diagram and its significance	Flight determination of drag polar- speed power method
S-8	SLO-1	Wave drag and methods to minimize it	Thrust available characteristics of different engines	Absolute ceiling and service ceiling	Takeoff performance – ground roll	Flight determination of drag polar- incremental drag method
	SLO-2	Solving Problems	Selection of Powerplant	Solving problems	Estimation of takeoff distance	Flight determination of drag polar- prop feathered sinks method
S-9	SLO-1	Drag polar of vehicles in supersonic speeds	Thrust required vs velocity- Graphical approach	Gliding flight- Equation of motion, glide hodograph	Landing performance – Estimation of landing distance	Flight determination of drag polar- incremental power method
	SLO-2	Drag polar of vehicles in hypersonic speeds	Thrust required vs velocity- Analytical approach	Minimum sink rate and minimum glide path angle conditions	Methods to minimise landing distance	Solving problems

Learning Resources	1. Perkins, C. D., and Hage, R. E., "Airplane Performance, Stability and Control," Wiley Toppan, 1974	3. Nelson, R.C., "Flight Stability and Automatic Control", McGraw Hill, 1989.
	2. John D. Anderson, "Aircraft Performance and Design", McGraw-Hill, 1999	4. McCormik, B. W., "Aerodynamics, Aeronautics and Flight Mechanics", John Wiley, 1995.

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry		Experts from Higher Technical Institutions
1. Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in		1. Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu
2. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com		2. Dr. P. K Dash, Nitte Meenakshi Institute of Technology, Bangalore, drpdash@gmail.com
		Internal Experts
		1. Mr. M. Vignesh Kumar, SRMIST
		2. Mr. K. Allwyn, SRMIST

Course Code	18ASC303J	Course Name	ROCKET PROPULSION	Course Category	C	Professional Core	L 3	T 0	P 2	C 4
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Pre-requisite Courses	18ASC203T	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards		Gas Tables	

Course Learning Rationale (CLR):		The purpose of learning this course is to:		Learning			Program Learning Outcomes (PLO)																
CLR-1 :	Understand the basic principles of rocket propulsion system	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	1	2	3	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15	
CLR-2 :	Understand the basic performance parameters of chemical propellants.				Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3				
CLR-3 :	Design the Solid Propellant Rocket				H	-	H	-	-	-	-	-	-	-	-	-	M	-	M				
CLR-4 :	Design the Liquid Propellant Rocket				H	H	H	-	-	-	-	-	-	-	-	-	M	M	M				
CLR-5 :	Understand the working of advanced rocket propulsion techniques				H	-	H	H	-	-	-	-	-	-	-	-	M	M	M				
CLR-6 :	Understand Rocket propulsion system, design, advanced propulsion system and its applications				H	H	H	H	-	-	-	-	-	-	-	-	M	M	M				
Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:																					
CLO-1 :	Design a multistage rocket and analyze its performance	2	85	75																			
CLO-2 :	Analyze the performance parameter and choose the chemical propellant	2	85	75																			
CLO-3 :	Analyze the Solid Propellant Rocket	2	85	75																			
CLO-4 :	Analyze the Liquid Propellant Rocket	2	85	75																			
CLO-5 :	Understand the working of advanced rocket propulsion techniques	2	85	75																			
CLR-6 :	Have a detailed knowledge of Rocket propulsion system, design, advanced propulsion system and its applications	2	85	75																			

Duration (hour)	15	15	15	15	15
S-1	SLO-1 SLO-2	History and evolution of rockets	Introduction of chemical propellant	Introduction of Solid Propellant Rocket	Introduction of Liquid Propellant Rocket
S-2	SLO-1 SLO-2	Rocket working principle	Molecular mass	Classifications of Solid Propellant Rocket	Classifications of Liquid Propellant Rocket
S-3	SLO-1 SLO-2	Rocket equation	specific heat ratio	Mechanism of burning	Propellant feed systems
S-4	SLO-1 SLO-2	Numerical Problems on Rocket equation	Stoichiometric ratio	Pressure feed system	Arc-jet thruster
S-5	SLO-1 SLO-2	Lab 1: Study of Piston Engines	Lab 4: Determination of convective heat transfer coefficient over a flat plate by forced convection	Lab 7: Performance test on a propeller	Resistojet thruster
S-6	SLO-1 SLO-2	Mass ratio of rocket	mixture ratio and equivalence ratio	Lab 10: Burning rate measurement of solid propellant by using Window bomb setup	Lab 13: Hybrid Rocket motor fuel grain preparation
S-7	SLO-1 SLO-2	Numerical Problems on rocket parameters	Energy release during combustion	Turbo pump feed system	Electro-static propulsion system
S-8	SLO-1 SLO-2	Rocket Nozzles and its Classifications	Heat of formation	Numerical Problems on feed system	Electro-magnetic propulsion system
S-9	SLO-1 SLO-2	Nozzle Performance	Heat of combustion	Injector	Ion thruster
S-10	SLO-1 SLO-2	Nozzle area ratio- Mass flow rate	Criterion for choice of propellant	Types of injector	Hall Effect Thruster
S-11	SLO-1 SLO-2	Numerical Problems on nozzle	Solid propellants	Numerical Problems on injector	Magneto plasma dynamic thruster
S-12	SLO-1 SLO-2	Lab 2: Study of jet engines	Lab 5: Characteristic plots of a free jet through a non-circular orifice	Lab 11: Thrust measurement on a prepared solid propellant	Pulsed Plasma Thruster
S-13	SLO-1 SLO-2		Lab 8: Thrust measurement of a Ramjet engine	Lab 14: Regression rate measurement of hybrid rocket motor	

Duration (hour)	15	15	15	15	15
S-11	SLO-1 Characteristic velocity and Thrust coefficient	Composition and processing	Burn rate index for stable operation	Thrust chamber	Nuclear rockets
	SLO-2 Numerical Problems on nozzle performance	Liquid propellants	Action time and burn time	Cooling of Thrust chamber	
S-12	SLO-1 Performance parameters and Efficiencies of rocket	Classifications of liquid propellant	Design of Solid Propellant rocket	Cryogenic propulsion system	Solar Propulsion system
	SLO-2 Numerical Problems on Performance parameters and Efficiencies of rocket	Storability of liquid propellant		Special features of cryogenic systems	
S-13	SLO-1 Staging and Clustering of rocket	Numerical Problems on chemical propellant	Numerical Problems on Solid Propulsion system	Numerical Problems on Liquid Propulsion system	Numerical problems
	SLO-2 Numerical Problems on Staging and Clustering of rocket				
S 14-15	SLO-1 Lab 3: Determination of convective heat transfer coefficient over a flat plate by natural convection	Lab 6: Characteristic plots of a wall jet through a non-circular orifice	Lab 9: Solid Rocket motor propellant preparation	Lab 12: Study of Liquid propulsion system	Lab 15: Thrust measurement of a hybrid rocket
	SLO-2				

Learning Resources	1. Ramamurthi.K, "Rocket propulsion", Laxmi Publications, India, Second edition 2016.	3. Philip Hill and Carl Peterson, "Mechanics and thermodynamics of propulsion", Pearson India, second edition 2010.
	2. George P. Sutton, Oscar Biblarz, "Rocket propulsion elements", Wiley India Pvt Ltd. eighth Edition 2010.	4. Stephen R. Turns, "An Introduction to Combustion: Concepts and Applications", McGraw-Hill Education, third Edition, 2011.

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember Understand	20 %	20 %	15 %	15 %	15 %	15 %	15 %	15 %	15 %	15 %
Level 2	Apply Analyze	20 %	20 %	20 %	20 %	20 %	20 %	20 %	20 %	20 %	20 %
Level 3	Evaluate Create	10 %	10 %	15 %	15 %	15 %	15 %	15 %	15 %	15 %	15 %
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. S. Raja, Senior Principal Scientist and Professor, NAL – Bangalore. raja@nal.res.in	1. Dr P K Dash Professor and HOD at Nitte Meenakshi Institute of Technology Bengaluru, Karnataka, India	1. Mr.G.Saravanan, SRMIST
	2. Dr.K.M.Parammasivam, Professor, MIT-Chennai. mparams@mitindia.edu	3. Mr. A. Vinoth Kumar, SRMIST

Course Code	18ASC304J	Course Name	AIRCRAFT STRUCTURES	Course Category	C	Professional Core	L	T	P	C
							3	0	2	4

Pre-requisite Courses	18ASC201J	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace	Data Book / Codes/Standards			

Course Learning Rationale (CLR):		The purpose of learning this course is to:			Learning		
CLR-1 :	Understand the physical meaning of symmetric bending and unsymmetrical bending				1	2	3
CLR-2 :	Know the procedure to find the shear flow and shear center in open sections subjected to shear loads				Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)
CLR-3 :	Know the procedure to find the shear flow and shear center in closed sections subjected to shear loads						
CLR-4 :	Know the procedure to find the shear flow in closed sections subjected to torque						
CLR-5 :	Understand the buckling analysis of plates						
CLR-6 :	Learn the experimental procedure of stress analysis in wing and fuselage						
Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:					
CLO-1 :	Explain the differences between symmetrical bending and unsymmetrical bending				2	85	75
CLO-2 :	Improve the ability in solving geometrical applications of a structure				2	85	75
CLO-3 :	Gain the ability to solve the shear flow analysis and shear centre of open section beams				2	85	75
CLO-4 :	Solve the shear flow and shear center calculations in closed sections subjected to both shear loads and torsion loads.				2	85	75
CLO-5 :	Describe the buckling modes of thin plates for various end conditions				2	85	75
CLO-6 :	Describe the stress analysis in wing and fuselage				2	85	75

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
H	H	H	H	-	H	H	H	-	-	-	M	M	M	M
H	H	H	H	-	H	H	H	-	-	-	M	M	M	M
H	H	H	H	-	H	H	H	-	-	-	M	M	M	M
H	H	H	H	-	H	H	H	-	-	-	M	M	M	M
H	H	H	H	H	H	H	H	-	-	-	M	M	M	M
H	H	H	H	H	H	H	H	-	-	-	M	M	M	M

Duration (hour)	15	15	15	15	15
S-1	SLO-1 Introduction to symmetrical bending	Introduction to thin walled beams	Concept of Bredt - Batho theory and assumptions for Engineers Theory of bending.	Introduction to thin plates and its application to aircraft structures.	Shear distribution for wings
	SLO-2 Concept of bending stresses in beams of symmetrical sections	Section properties of thin walled beams	Derivation of the relation between shear flow and the torque	Plate subjected to pure bending.	Numerical solving
S-2	SLO-1 Symmetrical bending, assumptions	Concept of shear flow and the shear centre.	Multi cell structure subjected to torque, shear flow determination	Flexural rigidity of a plate, anticlastic and synclastic surface	Bending moment distribution for wings
	SLO-2 Direct stress distribution	Introduction to thin walled open section beams, assumptions	Numerical solving	Plate subjected to bending and twisting	Numerical solving
S-3	SLO-1 Anticlastic bending, unsymmetrical bending, sign conventions and notation, Resolution of bending moments.	General stress, strain and displacement relationship for open thin-walled beams.	Shear flow and shear centre determination, and shear flow distribution for a thin-walled symmetrical (three cell) closed sections subjected to torque.	Determination of shear strain for thin plates	Shear distribution for fuselage
	SLO-2 Derivation for bending stress expression due to bending.	Shear flow expression for open sections.	Numerical solving	Plates subjected to a distributed transverse load	Numerical solving
S-4	SLO-1 Lab 1: Determination of the position of principal axes of a Z section	Lab 4: Verification of principle of superposition (Cantilever beam)	Lab 7: Determination of shear center of closed section beam	Lab 10: Determination of the buckling stress of a given column	Lab 13: Study of free and forced vibration analysis
	SLO-2 Position of neutral axis, calculation of section properties, Second moments of	Shear flow of C section	Shear flow and shear centre determination, and shear flow distribution for a thin-walled	Plate element subjected to bending, twisting and transverse loads.	Bending moment distribution for fuselage



Duration (hour)		15	15	15	15	15
		area of an inclined thin section and a semicircular section		symmetrical (single cell) closed sections subjected to shear force		
	SLO-2	Approximations for thin-walled sections	Shear flow of C section	Derivation of shear flow for closed section	Numerical solving	Numerical Solving
S-6	SLO-1	Bending of symmetric sections with symmetric loads - Numericals	Shear center of C section	Closed section single cell -shear flow determination	Boundary conditions for various types of edge supports	Shear resistant web beams.
	SLO-2	Numerical solving	Procedure of solving shear centre of c section	Numerical solving	Numerical solving	Numerical solving
S-7	SLO-1	Bending of symmetric sections with skew loads – Problems.	Shear flow distribution for thin walled open sections	Closed section two cell structure-shear flow determination	Combined bending and in-plane loading of a thin rectangular plate	Typical wing structural arrangement, wing strength requirements.
	SLO-2	Numerical solving	Shear center position for different type of thin walled open section	Numerical solving	Numerical solving	Numerical solving
S-8	SLO-1	Lab 2: Verification of Maxwell's reciprocal theorem(Cantilever beam)	Lab 5: Verification of principle of superposition (Simply supported beam)	Lab 8: Analysis of constant strength beam	Lab 11: Determination of the ratio of forces carried by two wires of different materials supported by hinged bar	Lab 14:Preparation of a composite laminate
	SLO-2					
S-9	SLO-1	Bending stress determination for symmetrical section with stringers.	Shear flow and shear centre determination, and shear flow distribution for thin-walled symmetrical open sections	Shear flow and shear centre determination, and shear flow distribution for a thin-walled symmetrical (single cell) closed sections subjected to shear force.	Buckling of thin plates	Tension (Wagner's) field beam – complete diagonal tension field beams.Derivation
	SLO-2	Numerical solving	Numerical solving	Numerical solving	Numerical solving	Incomplete diagonal tension field beams
S-10	SLO-1	Bending of unsymmetric sections with skew loads - Problems	Shear flow and shear centre determination, and shear flow distribution for thin-walled unsymmetrical open sections	Shear flow and shear centre determination, and shear flow distribution for a thin-walled symmetrical (two cell) closed sections subjected to shear force.	Inelastic buckling of plates	Typical wing structural arrangement, wing strength requirements.
	SLO-2	Numerical solving	Numerical solving	Numerical solving	Numerical solving	Beam theory assumptions, wing stress analysis methods.
S-11	SLO-1	Bending stress determination for unsymmetrical section with stringers.	Concept of structural idealization.	Shear flow and shear centre determination, and shear flow distribution for symmetrical and unsymmetrical (single cell) closed sections with stringers subjected to shear force	Local instability and Instability of stiffened panels	Shear lag concepts
	SLO-2	Numerical solving	Determination of boom areas	Numerical solving	Numerical solving	Basic fuselage structure, fuselage stress analysis methods.
S-12	SLO-1	Lab 3: Verification of Maxwell's reciprocal theorem(Simply supported beam)	Lab 6: Determination of shear center of open section beam	Lab 9: Determination of principal axes of a hollow shaft subjected to both bending and torsional loads	Lab 12:Determination of hoop stress and longitudinal stress in a thin walled pressure vessel	Lab 15:Design aspect of tension field beam
	SLO-2					
S-13	SLO-1	Bending of thin symmetric sections with symmetric loads - Problems	Shear flow and shear centre determination, and shear flow distribution for symmetrical open sections with stringers.	Shear flow and shear centre determination, and shear flow distribution for symmetrical and unsymmetrical (two cells) closed sections with stringers subjected to shear force.	Flexural–torsional buckling of thin-walled columns	shear flow distribution in the web of the tapered beam
	SLO-2	Numerical solving	Numerical solving	Numerical solving	Numerical solving	Calculation of direct stress due to bending in a fuselage

Duration (hour)		15	15	15	15	15
S-14	SLO-1	Bending of thin unsymmetric sections with skew loads - Problems.	Shear flow and shear centre determination, and shear flow distribution for a unsymmetrical open sections with stringers.	Shear flow and shear centre determination, and shear flow distribution for symmetrical and unsymmetrical (three cells) closed sections with stringers subjected to shear force.	Estimation of crippling stress using Needham's and Gerard's method.	shear flow distribution in the fuselage
	SLO-2	Numerical solving	Numerical solving	Numerical solving	Stiffened panel / Sheet effective width concepts	Principles of stiffener / web construction
S-15	SLO-1	Bending of thin symmetric sections with skew loads - Problems.	Walls effective in bending	Walls effective in bending	Inter rivet and sheet wrinkling failures.	Fuselage frames
	SLO-2	Numerical solving	Walls ineffective in bending	Walls ineffective in bending	Thin walled column strength, Torsional instability of thin walled columns.	Wing ribs

Learning Resources	1. Megson T H G, 'Aircraft Structures for Engineering Students', Elsevier, Fifth edition, 2013 2. Bruhn. E.F., 'Analysis and Design of Flight Vehicles Structures', Tri-state offset company, USA 1985 3. Aircraft Structures Laboratory manual	4. Rivello, R.M., Theory and Analysis of Flight Structures, McGraw Hill, 1993. 5. Peery, D.J., Aircraft Structures, 2nd edition, McGraw-Hill, N.Y., 1999
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	20%	20%	15%	15%	15%	15%	15%	15%	15%	15%
	Understand										
Level 2	Apply	20%	20%	20%	20%	20%	20%	20%	20%	20%	20%
	Analyze										
Level 3	Evaluate	10%	10%	15%	15%	15%	15%	15%	15%	15%	15%
	Create										
	Total	100%		100%		100%		100%		100%	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. G.Balamurugan, National Aerospace Laboratories, Bangalore, gbala@nal.res.in	1. Dr. V.Arumugam, Madras Institute of Technology, Chennai, arumugam.mitaero@gmail.com	1. Dr.LR.Ganapathy Subramanian, Professor,SRMIST
2. Dr.A. Sakthivel, CEMILAC, Bangalore, asakthironika@gmail.com	2. Dr. R. Velmurugan, Indian Institute of Technology Madras, rvel@ae.iitm.ac.in	2. Mr. S.Chandra Sekhar Assistant Professor,SRMIST

Course Code	18ASC305T	Course Name	FLIGHT DYNAMICS - II	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	18ASC302T	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards			

Course Learning Rationale (CLR):	The purpose of learning this course is to:	Learning			Program Learning Outcomes (PLO)														
CLR-1 :	Know the importance of stability and control of airplane	1	2	3	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
CLR-2 :	Learn the concepts of static stability and dynamic stability of airplane	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
CLR-3 :	Learn longitudinal, lateral and directional stabilities				H	M	M	H	L	-	H	-	-	-	-	H	H	H	-
CLR-4 :	Understand control fixed and control free effects				M	M	M	M	L	-	H	-	M	-	-	H	H	H	-
CLR-5 :	Learn the criteria for stability and instability				M	H	M	H	M	-	H	-	-	-	-	H	H	H	-
CLR-6 :	Learn experimental techniques to measure stability parameters of an airplane using real flight testing				M	M	M	M	L	-	L	-	M	-	-	H	H	H	M
					M	M	H	H	M	M	H	M	H	-	M	H	H	H	-
					M	H	M	M	M	-	-	-	M	M	-	H	-	H	M
Course Learning Outcomes (CLO):	At the end of this course, learners will be able to:																		
CLO-1 :	Determine the degree of stability of any airplane configuration	5	85	90															
CLO-2 :	Design an airplane for the required degree of stability and maneuverability	3	80	85															
CLO-3 :	Indicate the different flight techniques for safer flight	4	75	80															
CLO-4 :	Perform basic calibration of control surfaces	5	50	75															
CLO-5 :	Identify different instability modes in flight and suggest the correct recovery procedure	5	80	80															
CLO-6 :	Determine the stability derivatives of a real airplane with real flight testing	4	00	95															

Duration (hour)		9	9	9	9	9
S-1	SLO-1	Basic concepts of stability and control	Longitudinal control- basics	Dynamic stability- basics	Construction of stability matrix	Lateral- directional equations of motion
	SLO-2	Classification of stability- static and dynamic	Various factors affecting the design of control surface	Rigid body equation of motion for a disturbed flight- Forces	Decoupling of longitudinal and lateral modes	Spiral approximation- spiral divergence
S-2	SLO-1	Longitudinal static stability – basic criteria	Control effectiveness- elevator effectiveness	Rigid body equation of motion for a disturbed flight- moment	Significance of eigen values and eigen vectors of stability matrix	Directional divergence
	SLO-2	Wing contribution to longitudinal static stability	Solving problems	Defining equations of motion in inertial frame of reference	Solving problems	Dutch roll approximation
S-3	SLO-1	Tail contribution (Aft tail contribution) to longitudinal static stability	Elevator angle to trim	Euler angles	Solving Problems	Routh's criterion
	SLO-2	Solving problems	Solving problems	Orientation and position of the airplane	Longitudinal motion-stick fixed	Wing stall characteristics
S-4	SLO-1	Fuselage contribution	Elevator hinge moment	Gravitational and thrust forces	Second order differential equations	Autorotation and its recovery
	SLO-2	Power plant contribution	Stick free neutral point- definition	Small disturbance theory- assumptions	Variation of response with damping ratio	Spin (stalled) and its recovery
S-5	SLO-1	Canard configuration (Forward tail surface) contribution to longitudinal static stability	Relation between stick fixed and stick free neutral point	Small disturbance theory-linearized longitudinal rigid body equation of motion	Pure pitching motion	Aeroelastic effects- wing torsional divergence
	SLO-2	Stick fixed neutral point- definition	Aerodynamic balancing- basics	Small disturbance theory-linearized lateral rigid body equation of motion	State variable representation of the equations of motion	Aeroelastic effects- control reversal
S-6	SLO-1	Solving problems	Various methods of aerodynamic balancing- Set back hinge, horn balance	Aerodynamic force and moment representation by stability coefficient	Solving problems	Aeroelastic effects- control surface flutter

Duration (hour)		9	9	9	9	9
	SLO-2	Basic criteria for longitudinal, roll and directional stability	Various methods of aerodynamic balancing- Frise aileron, sealed nose balance and servo tab	Derivatives due to the change in forward speed	Longitudinal approximations	Stick fixed neutral point
S-7	SLO-1	Contribution of various components to roll stability	Trim tabs	Derivatives due to the change in pitching velocity	Phugoid and short period oscillations	Determination of stick fixed neutral point-parameters
	SLO-2	Dihedral effect and roll control	Critical conditions of stability and control	Derivatives due to the time rate of change of angle of attack	Influence of stability derivative on the longitudinal modes of motion	Determination of stick free neutral point-parameters
S-8	SLO-1	Solving problems	Static margin and neutral point- definition	Derivatives due to the rolling rate	Lateral motion – stick fixed	Determination of stick fixed maneuvering point- parameters
	SLO-2	Directional stability- basics	Stick forces and stick force gradients- definition	Derivatives due to the yawing rate	Pure rolling motion	Determination of stick free maneuvering point- parameters
S-9	SLO-1	Various requirements of rudder for directional control	Solving problems	Solving problems	Tutorial	Tutorial
	SLO-2	Solving problems	Solving problems	Solving problems	Tutorial	Tutorial

Learning Resources	1. Nelson, R.C., "Flight Stability and Automatic Control", McGraw Hill, 1989 2. Bernard Etkin "Dynamics of atmospheric flight" Wiley, 1972	3. Perkins, C. D., and Hage, R.E., "Airplane Performance, Stability and Control," Wiley Toppan, 1974.. 4. Babister, A. W., "Aircraft Stability and Response", Pergamon Press, 1980. 5. L J Clancy "Aerodynamics" John Wiley & Sons (1975)
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

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Course Designers											
Experts from Industry						Experts from Higher Technical Institutions				Internal Experts	
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2. Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com						2. Dr. P. K Dash, Nitte Meenakshi Institute of Technology, Bangalore, drpdash@gmail.com				2. Mr. K. Allwyn, SRMIST	



Course Code	18ASC306T	Course Name	INTRODUCTION TO SPACE TECHNOLOGY	Course Category	C	Professional Core	L	T	P	C
							3	0	0	3

Pre-requisite Courses	18ASC101T, 18MAB102T	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards			

Course Learning Rationale (CLR):		The purpose of learning this course is to:			Learning			Program Learning Outcomes (PLO)														
		1	2	3	1	2	3	4	5	6	7	8	9	10	11	12	13	14	15			
CLR-1 :	Learn about fundamental laws that govern the orbital dynamics.	Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3			
CLR-2 :	Understand and apply the basic equations of orbital dynamics for different conic orbits.				H	M	L	L	-	-	-	-	-	-	L	M	M	M				
CLR-3 :	Understand the importance of Keplerian orbital elements, Kepler's equation and orbital perturbations.				H	H	M	M	L	-	-	-	-	-	-	L	H	H	M			
CLR-4 :	Learn the different types of orbit transfers, basics of interplanetary trajectories.				H	M	M	M	-	-	-	-	-	-	-	L	M	M	H			
CLR-5 :	Study the governing equations of rocket motion, rocket motion under different conditions.				H	H	M	M	L	-	-	-	-	-	-	L	H	M	M			
CLR-6 :	Understand the fundamentals of ballistic missile trajectory and its governing equations, errors associated during launching of ballistic missiles.				H	M	L	L	-	-	-	-	-	-	-	L	M	H	M			
Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:			Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)	H	M	M	L	-	-	-	-	L	M	M	M			
CLO-1 :	Acquire knowledge of laws applied in orbital dynamics.	2	85	75	H	M	M	L	-	-	-	-	-	-	-	L	M	M	M			
CLO-2 :	Analyze the properties of conic orbits using the governing equations of orbital mechanics.	2	90	80	H	H	M	M	L	-	-	-	-	-	-	L	H	H	M			
CLO-3 :	Apply Kepler's equation, earth's oblateness for positioning the satellite at a desired position in orbit.	2	80	75	H	M	M	M	-	-	-	-	-	-	-	L	M	M	H			
CLO-4 :	Accrue knowledge of different orbit transfers which can be used for many practical missions.	2	85	75	H	H	M	M	L	-	-	-	-	-	-	L	H	M	M			
CLO-5 :	Apply the governing equations of rocket motion for different conditions.	2	80	70	H	M	L	L	-	-	-	-	-	-	-	L	M	H	M			
CLO-6 :	Accrue understanding and importance of ballistic missile trajectory and effectively implement its equations by considering the launching errors.	2	85	75	H	M	M	L	-	-	-	-	-	-	-	L	M	M	M			

Duration (hour)	9	9	9	9	9
S-1	SLO-1 Laws of Orbital Dynamics - Newton's Laws of Motion	Introduction to Coordinate Systems	Introduction to Orbital Maneuver, Impulsive Maneuvers	Rockets – Introduction, Tsiolkov sky rocket equation, Rocket Parameters	Introduction Ballistic missile and its trajectory
S-2	SLO-2 Kepler's Laws, Conic Sections	Types of Coordinate Systems			
S-2	SLO-1 Structure of Upper Atmosphere	Time Systems - Definitions	Introduction to Single Impulse Maneuver – Adjustment of Heights, Simple Rotation.	Rocket Motion – Introduction, Two Dimensional Rocket Motion	Free-Flight Range Equation - Derivation
S-2	SLO-2 Multi-Body / N-body problem	Orbital Elements - Introduction			
S-3	SLO-1 Two Body Problem – Introduction	Classical / Keplerian Orbital Elements	Introduction to Single Impulse Maneuver – Simple Plane Change, Combined Change of Plane and Height	Two Dimensional Equations of Motion of Rigid Rockets – Assumptions, Derivation	Flight-Path Angle Equation – Derivation
S-3	SLO-2 Two Body Equation of Relative Motion – Assumptions, Significance		Introduction to Hohmann Transfer		
S-4	SLO-1 Orbit / Trajectory Equation	Kepler's Equation – Derivation	Hohmann Transfer	Rocket motion in Free Space	Numerical Problems on Free-Flight Range and Flight-Path Angle Equations
S-5	SLO-2 Constants of Motion – Introduction	Numerical Problems on Kepler's Equation	Bi-elliptic Transfer	Rocket Motion in Homogeneous Gravitational Field	Maximum Range Trajectory
S-5	SLO-1 Conservation of Specific Angular Momentum				Time of Free-Flight
S-6	SLO-1 Conservation of Specific Mechanical Energy – Vis-Viva Equation	Types of Satellite Orbits	Numerical Problems on Hohmann and Bi-elliptic Transfers	Introduction to Vertical Flight Trajectory	Effect of Earth Rotation - Compensating for the Initial velocity

Duration (hour)	9	9	9	9	9
SLO-2					Numerical Problems on Initial Velocity Compensation
S-7	SLO-1	Relating the Constants of Motion to the Geometry of the Orbit	Earth's Oblateness and its Effects, Applications – Sun-Synchronous and Molniya Orbits	One Tangent Burn Transfer - Introduction	Introduction to Constant Pitch Angle and Gravity Turn / Zero-Lift Trajectories
	SLO-2	Proof - Kepler's Second and Third Law		Phasing Maneuvers - Introduction	
S-8	SLO-1	Some Important Properties of Individual Conic Orbits – Circular Orbit, Elliptic Orbit	Introduction to Orbital Perturbations – Perturbing Forces, Perturbation Techniques – Definitions of Special and General Perturbations	Introduction to Inter-planetary Mission Trajectories	Introduction to Multi-Stage Rocket and its Types, Restricted Staging and Optimal Staging - Definitions
	SLO-2	Some Important Properties of Individual Conic Orbits – Parabolic Orbit, Hyperbolic Orbit			
S-9	SLO-1	Tutorial	Tutorial	Tutorial	Tutorial
	SLO-2				

Learning Resources	<ol style="list-style-type: none"> <li>Howard D. Curtis, <i>Orbital Mechanics for Engineering Students</i>, 4<sup>th</sup> Edition, Butterworth-Heinemann, 2019.</li> <li>William E. Wiesel, <i>Spaceflight Dynamics</i>, 3<sup>rd</sup> Edition, CreateSpace, 2010.</li> <li>Roger R. Bate, Donald D. Mueller &amp; Jerry E. White, <i>Fundamentals of Astrodynamics</i>, Dover Publications, Inc., New York, 1971.</li> </ol>	<ol style="list-style-type: none"> <li>Ashish Tewari, <i>Atmospheric and Space Flight Dynamics</i>, Springer, 2007.</li> <li>Cornelisse J.W., Schoyer H.F.R. &amp; Wakker K.F., <i>Rocket Propulsion and Spaceflight Dynamics</i>, Pitman Publishing Ltd., 1979.</li> <li>Martin J. L. Turner, <i>Rocket and Spacecraft Propulsion</i>, 3<sup>rd</sup> Edition, Springer, 2009.</li> <li>Vladimir A. Chobotov, <i>Orbital Mechanics</i>, 3<sup>rd</sup> Edition, AIAA Education Series, AIAA, 2002.</li> </ol>
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Understand										
Level 2	Apply	40 %	-	40 %	-	40 %	-	40 %	-	40%	-
	Analyze										
Level 3	Evaluate	20 %	-	30 %	-	30 %	-	30 %	-	30%	-
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1. Dr. Manishankar C., Senior Scientist, NAL, Bangalore	1. Prof. Arun Kumar P., Assistant Professor, IIT Jammu	1. Dr. S. M. Aravindh Kumar, SRMIST
		2. Mr. K. Allwyn, SRMIST

Course Code	18ASC307L	Course Name	AIRCRAFT COMPONENT DRAWING AND COMPUTATIONAL ANALYSIS LABORATORY	Course Category	C	Professional Core				L	T	P	C
										0	0	2	1

Pre-requisite Courses	18ASC201J, 18ASC301J	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering		Data Book / Codes/Standards	Gas Table	

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1 :	To create 2D Design of aircraft components			
CLR-2 :	To create 3-Dimensional Design and assembly of typical aircraft & its components.			
CLR-3 :	To create drafting of aircraft components			
CLR-4 :	To simulate and evaluate Structural characteristics of beams and aircraft structural components			
CLR-5 :	To simulate and evaluate aerodynamic properties of Subsonic and supersonic flow over the objects			
CLR-6 :	To simulate and evaluate Heat transfer and combustion process			

Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:		
CLO-1 :	To familiarize with basic aircraft components			
CLO-2 :	To familiarize 3-Dimensional Design of typical aircraft & its components.			
CLO-3 :	To familiarize assembly and drafting of aircraft components			
CLO-4 :	To familiarize Structural analysis of beams and aircraft structural components			
CLO-5 :	To familiarize Subsonic and supersonic flow analysis over the objects			
CLO-6 :	To familiarize Heat transfer and combustion analysis			

Learning		
1	2	3
Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)
1	80	75
3	80	75
3	70	60
3	70	60
3	70	60
3	70	60
3	70	60

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Engineering Knowledge	Problem Analysis	Design & Development	Analysis, Design, Research	Modern Tool Usage	Society & Culture	Environment & Sustainability	Ethics	Individual & Team Work	Communication	Project Mgt. & Finance	Life Long Learning	PSO - 1	PSO - 2	PSO - 3
H	L	M	M	H	-	-	-	-	-	-	H	-	H	-
H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
H	H	H	H	H	-	-	-	-	-	-	H	-	H	-
H	H	H	H	H	-	-	-	-	-	-	H	-	H	-

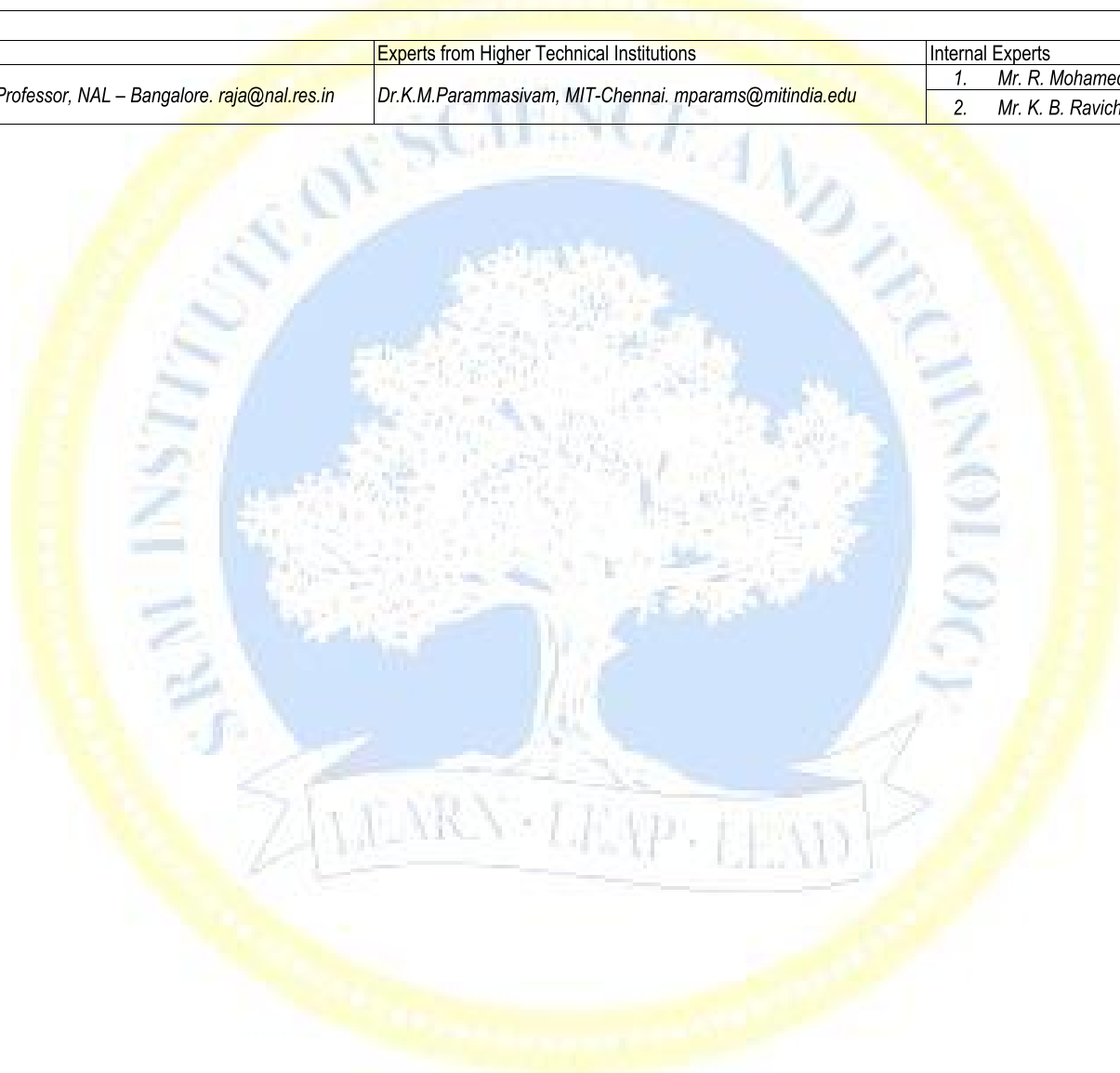
Duration (hour)		06	06	06	06	06
S 1-2	SLO-1 SLO-2	Lab 1: 2D layout of aircraft wing rib and bulkhead sections	Lab 4: Assembly of Typical Aircraft.	Lab 7: 2D analysis of subsonic flow over bluff /streamlined body.	Lab 10: Simulation of Premixed / Non-Premixed Combustion analysis	Lab 13: Heat transfer analysis over a flat plate with natural / forced convection
S 3-4	SLO-1 SLO-2	Lab 2: 3D model of aircraft Wing Structure	Lab 5: Drafting of Typical Aircraft.	Lab 8: 2D analysis of supersonic flow over bluff /streamlined body.	Lab 11: structural analysis of beams	Lab 14: Heat transfer analysis over a composite plate with natural / forced convection
S 5-6	SLO-1 SLO-2	Lab 3: 3D model of aircraft fuselage Structure	Lab 6: Introduction to CFD and Grid independency study	Lab 9: 2D analysis of supersonic flow through C-D Nozzle.	Lab 12: Structural analysis of aircraft wing	Lab 15: Heat Transfer analysis in a Thin Plate using MATLAB code

Learning Resources	Laboratory manual User manual of respective software	Gas Table
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Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (50% weightage)								Final Examination (50% weightage)	
		CLA – 1 (10%)		CLA – 2 (15%)		CLA – 3 (15%)		CLA – 4 (10%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	-	40%	-	30%	-	30%	-	30%	-	30%
	Understand										
Level 2	Apply	-	40%	-	40%	-	40%	-	40%	-	40%
	Analyze										
Level 3	Evaluate	-	20%	-	30%	-	30%	-	30%	-	30%
	Create										
	Total	100 %		100 %		100 %		100 %		100 %	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
Dr. S. Raja, Senior Principal Scientist and Professor, NAL – Bangalore. <a href="mailto:raja@nal.res.in">raja@nal.res.in</a>	Dr.K.M.Parammasivam, MIT-Chennai. <a href="mailto:mparams@mitindia.edu">mparams@mitindia.edu</a>	1. Mr. R. Mohamed Arif, SRMIST.
		2. Mr. K. B. RavichandraKumar, SRMIST.





Course Code	18ASC350T	Course Name	COMPREHENSION	Course Category	C	Professional Core	L	T	P	C
							0	1	0	1

Pre-requisite Courses	Nil	Co-requisite Courses	Nil	Progressive Courses	Nil
Course Offering Department	Aerospace Engineering	Data Book / Codes/Standards		Machine Data	

Course Learning Rationale (CLR):		The purpose of learning this course is to:		
CLR-1 :	Acquire skills to solve real world problems in Engineering Graphics Design, Engineering Mechanics and Fluid Mechanics			
CLR-2 :	Acquire skills to solve problems in Thermodynamics, Aircraft Systems and Instruments and Aircraft Materials, Production Techniques			
CLR-3 :	Acquire skills to solve real world problems in Applied Solid Mechanics, Incompressible Aerodynamics and Air Breathing Propulsion			
CLR-4 :	Acquire skills to solve real world problems in Compressible Aerodynamics, Rocket Propulsion and Aircraft Structures			
CLR-5 :	Acquire skills to solve real world problems for competitive examinations in Aerospace Engineering			
CLR-6 :	Acquire skills to solve real world problems in the broad domain of Aerospace Engineering			

Course Learning Outcomes (CLO):		At the end of this course, learners will be able to:		
CLO-1 :	Practice and gain confidence, competence to solveproblems in Engineering Graphics Design, Engineering Mechanics and Fluid Mechanics			
CLO-2 :	Practice and gain confidence, competence in Thermodynamics, Aircraft Systems and Instruments, Aircraft Materials, Production Techniques			
CLO-3 :	Solve problems in Applied Solid Mechanics, Incompressible Aerodynamics and Air Breathing Propulsion			
CLO-4 :	Practice and gain confidence and competence to solve problems in Compressible Aerodynamics, Rocket Propulsion and Aircraft Structures			
CLO-5 :	Practice and gain confidence and competence to solveproblems for competitive examinations in Aerospace Engineering			
CLO-6 :	Practice and gain confidence and competence to solve problems in the broad domain of Aerospace Engineering			

Learning		
1	2	3
Level of Thinking (Bloom)	Expected Proficiency (%)	Expected Attainment (%)

Program Learning Outcomes (PLO)														
1	2	3	4	5	6	7	8	9	10	11	12	13	14	15
Engineering Knowledge														
Problem Analysis														
Design & Development														
Analysis, Design, Modern Tool Usage														
Society & Culture														
Environment & Ethics														
Individual & Team Work														
Communication														
Project Mgt. & Finance														
Life Long Learning														
PSO - 1														
PSO - 2														
PSO - 3														

Duration (hour)		3	3	3	3	3
S-1	SLO-1	Tutorial on Engineering graphics and design	Tutorial on Thermodynamics	Tutorial on Applied Solid Mechanics	Tutorial on Compressible Aerodynamics	Problem Solving
	SLO-2	Problem Solving	Problem Solving	Problem Solving	Problem Solving	Problem Solving
S-2	SLO-1	Tutorial on Engineering Mechanics	Review of Aircraft Systems	Tutorial on Incompressible Aerodynamics	Tutorial on Rocket Propulsion	Problem Solving
	SLO-2	Problem Solving	Review of Aircraft Instruments	Problem Solving	Problem Solving	Problem Solving
S-3	SLO-1	Tutorial on Fluid mechanics	Review of Aircraft Materials	Tutorial on Air Breathing Propulsion	Tutorial on Aircraft Structures	Problem Solving
	SLO-2	Problem Solving	Review of Production Techniques	Problem Solving	Problem Solving	Problem Solving

Learning Resources	1. Ferdinand P. Beer, E. Russell Johnston Jr., David Mazurek, Philip J Cornwell, "Vector Mechanics for Engineers: Statics and Dynamics", McGraw - Hill, New Delhi, Tenth Edition, 2013.	4. Cohen. H. Rogers. G.F.C. and Saravanamuttoo. H.I.H.: Gas turbine theory. 4 <sup>th</sup> ed., Pearson
	2. Irving H. Shames, "Mechanics of Fluids", 4 <sup>rd</sup> Edition, McGraw-Hill, 2003.	5. George P. Sutton, Oscar Biblarz, "Rocket propulsion elements", Wiley India Pvt Ltd. 8 <sup>th</sup> ed., 2010
	3. Yunus A. Cengel and Michael A. Boles, "Thermodynamics an engineering approach", 7 <sup>th</sup> ed., , McGraw Hill, 2011	6. Anderson, J.D., "Fundamentals of Aerodynamics", McGraw Hill., New York, 6 <sup>th</sup> ed., 2016

Learning Assessment											
	Bloom's Level of Thinking	Continuous Learning Assessment (100% weightage)								Final Examination	
		CLA – 1 (20%)		CLA – 2 (30%)		CLA – 3 (30%)		CLA – 4 (20%)#			
		Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice	Theory	Practice
Level 1	Remember	40%	-	30%	-	30%	-	30%	-	-	-
	Understand										
Level 2	Apply	40%	-	40%	-	40%	-	40%	-	-	-
	Analyze										
Level 3	Evaluate	20%	-	30%	-	30%	-	30%	-	-	-
	Create										
	Total	100 %		100 %		100 %		100 %		-	

# CLA – 4 can be from any combination of these: Assignments, Seminars, Tech Talks, Mini-Projects, Case-Studies, Self-Study, MOOCs, Certifications, Conf. Paper etc.,

Course Designers		
Experts from Industry	Experts from Higher Technical Institutions	Internal Experts
1) Dr. Raja S, CSIR-NAL, Bangalore, raja@nal.res.in	1) Dr. K. M. Parammasivam, Madras Institute of Technology, Chennai, mparams@mitindia.edu	1) Mr.G.Mahendra Perumal , SRMIST
2) Wg.Cdr K.Manoharan (Retd), Blue Dart Aviation Ltd., manoharank@bluedart.com	2) Dr. P. K Dash, Nitte Meenakshi Institute of Technology, Bangalore, drpdash@gmail.com	2) Dr.S.Gurusideswar , SRMIST